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Nonlinear Network Structures for
Feedback Control



<http://ARRI.uta.edu/acs>





香港中文大學
The Chinese University of Hong Kong

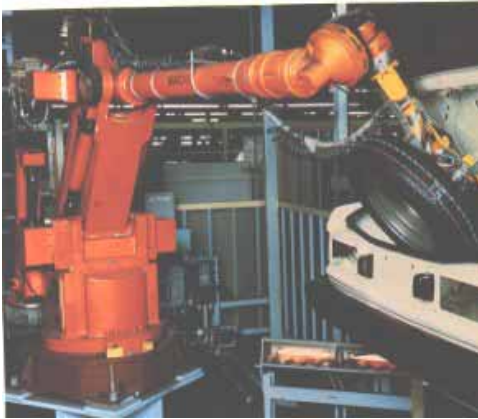


Organized and
invited by Professor
Jie Huang, CUHK

SCUT / CUHK Lectures
on Advances in Control
March 2005

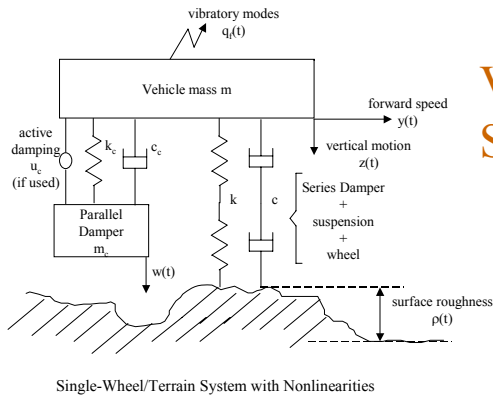
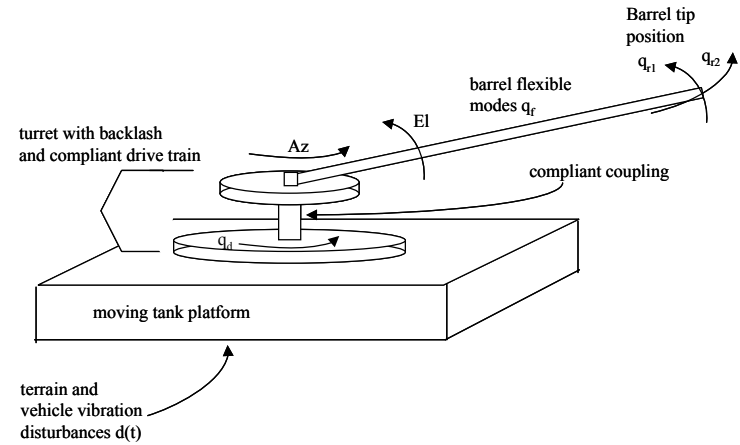
Relevance- Machine Feedback Control

High-Speed Precision Motion Control with unmodeled dynamics, vibration suppression, disturbance rejection, friction compensation, deadzone/backlash control



Industrial
Machines

Military Land
Systems



Vehicle
Suspension

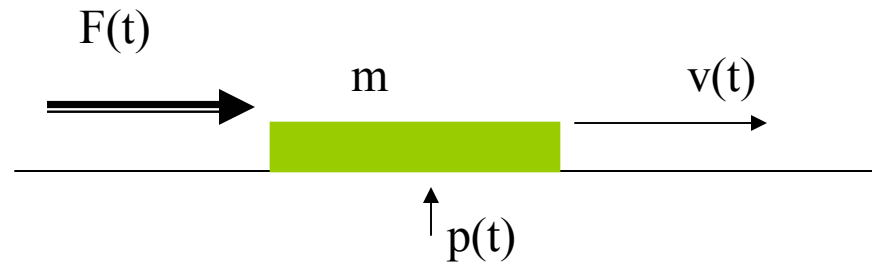
Aerospace



Newton's Law

$$F = ma = m\ddot{x}$$

$$\ddot{x} = \frac{F(t)}{m} \equiv u(t)$$



LaGrange's Eqs. Of Motion \implies

Mechanical Motion Systems (Vehicles, Robots)

$$M(\dot{q})\ddot{q} + V_m(q, \dot{q})\dot{q} + G(q) + F(\dot{q}) + \tau_d = B(q)\tau$$

inertia

Coriolis/centripetal
force

gravity

friction

disturbances

Actuator
problems

Control Input

Darwinian Selection & Population Dynamics

Volterra's fishes

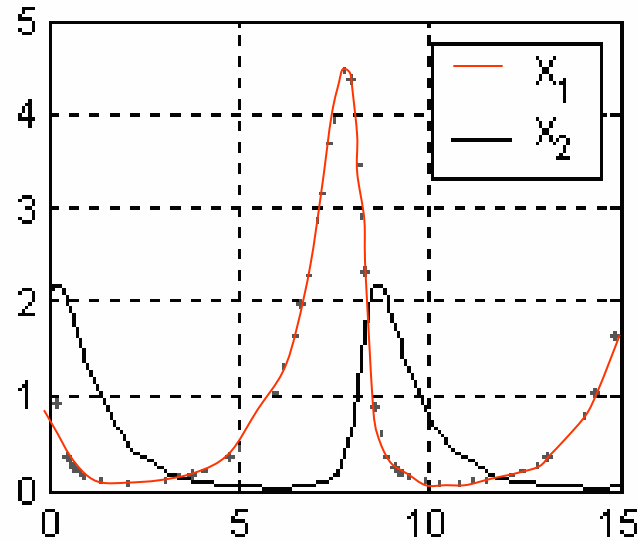
$$\dot{x}_1 = ax_1 - bx_1x_2$$

$$\dot{x}_2 = -cx_2 + dx_1x_2$$

x_1 = prey

x_2 = predator

Time Trajectory with $(x_{10}, x_{20}) = (2, 2)$



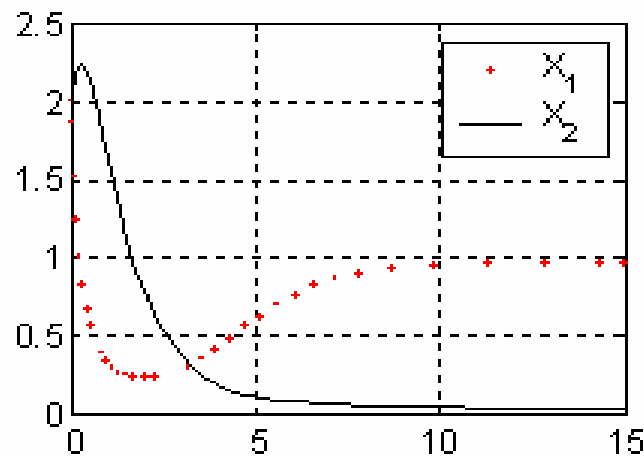
Stable
Limit Cycle

Effects of Overcrowding
Limited food and resources

$$\dot{x}_1 = ax_1 - bx_1x_2 - ex_1^2$$

$$\dot{x}_2 = -cx_2 + dx_1x_2$$

Favorable to Prey!



Stable
Equilibrium POINT

Dynamical System Models

Continuous-Time Systems

Discrete-Time Systems

Nonlinear system

$$\dot{x} = f(x) + g(x)u$$

$$y = h(x)$$

$$x_{k+1} = f(x_k) + g(x_k)u_k$$

$$y_k = h(x_k)$$

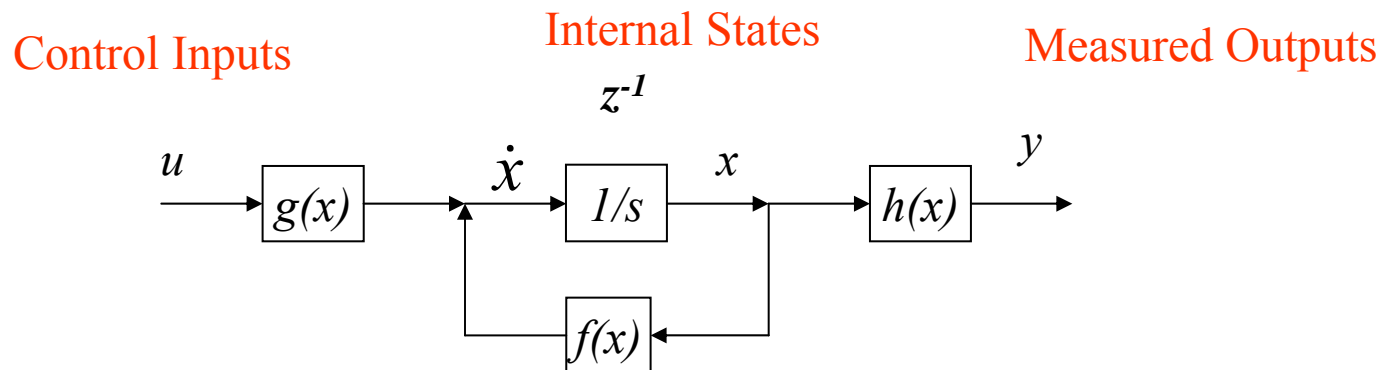
Linear system

$$\dot{x} = Ax + Bu$$

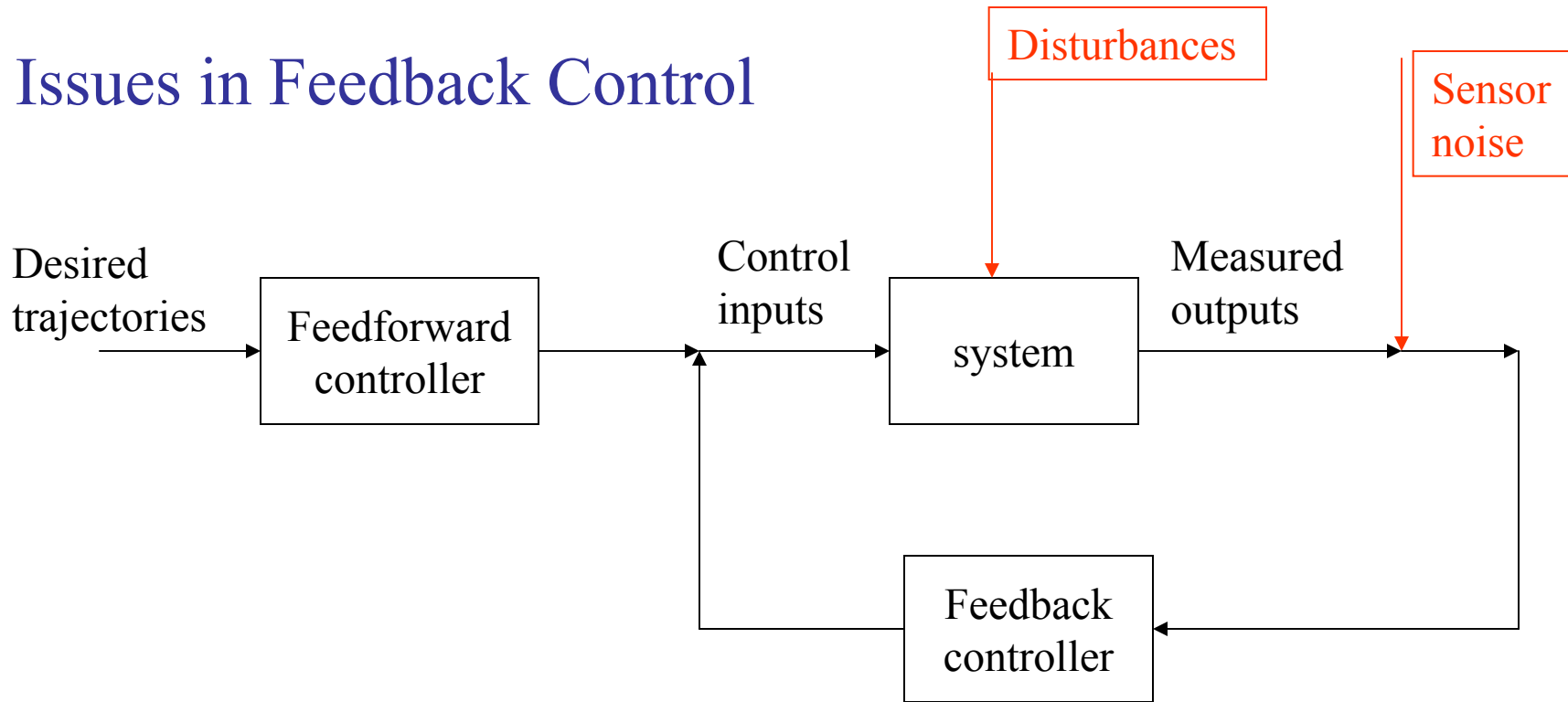
$$y = Cx$$

$$x_{k+1} = Ax_k + B_k$$

$$y_k = Cx_k$$



Issues in Feedback Control

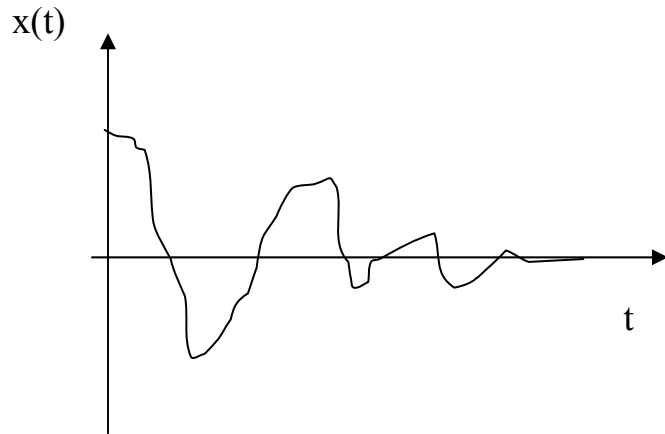


- Stability
- Tracking
- Boundedness
- Robustness
 - to disturbances
 - to unknown dynamics

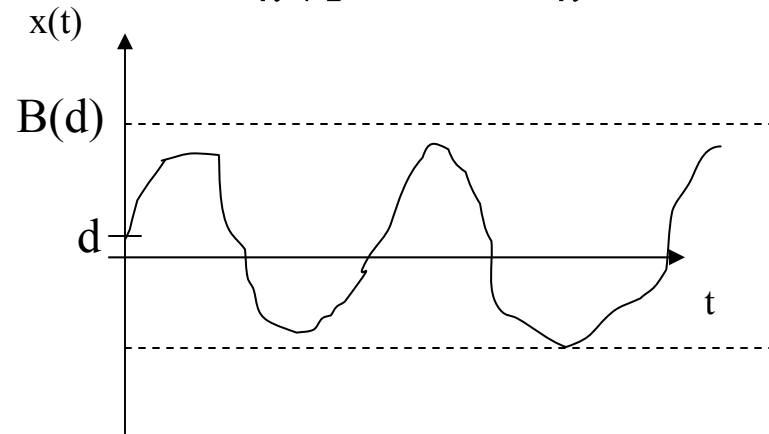
Definitions of System Stability

$$\dot{x} = f(x)$$

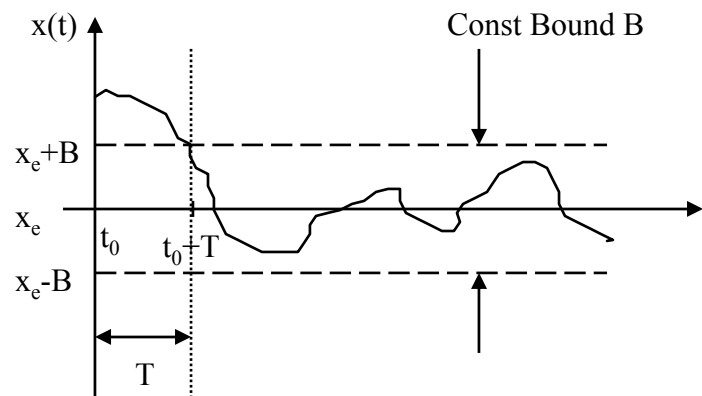
$$x_{k+1} = f(x_k)$$



Asymptotic Stability

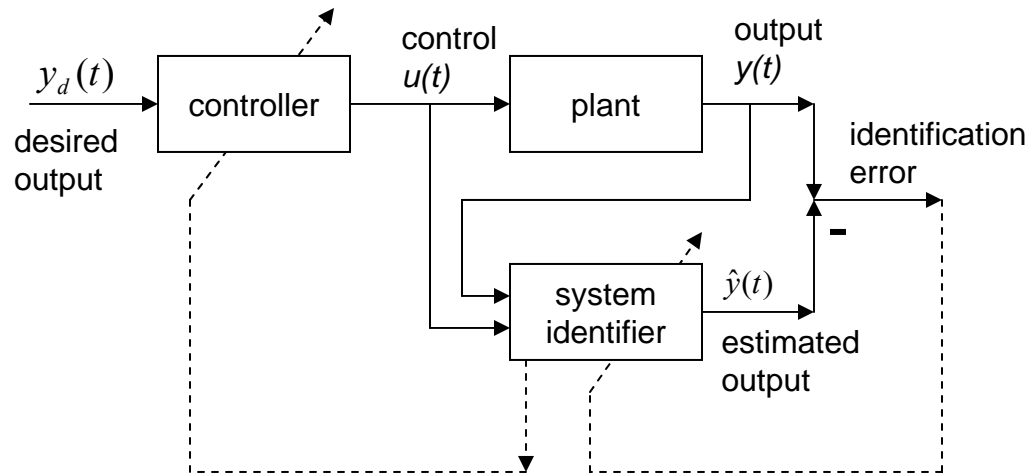


Marginal Stability

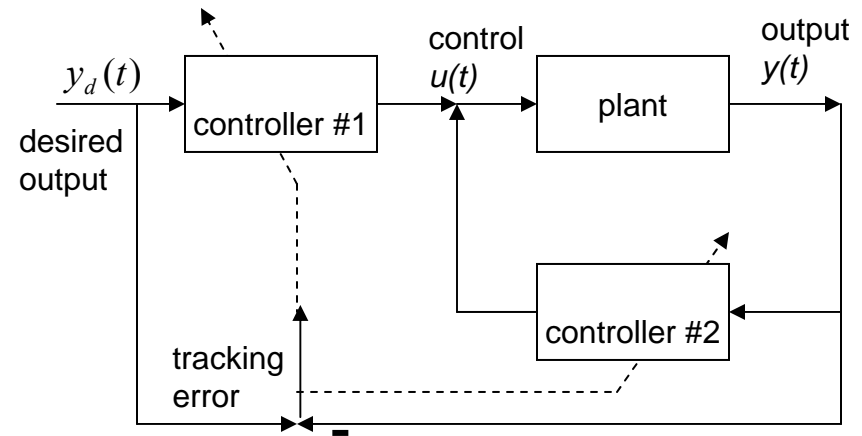


Uniform Ultimate Boundedness

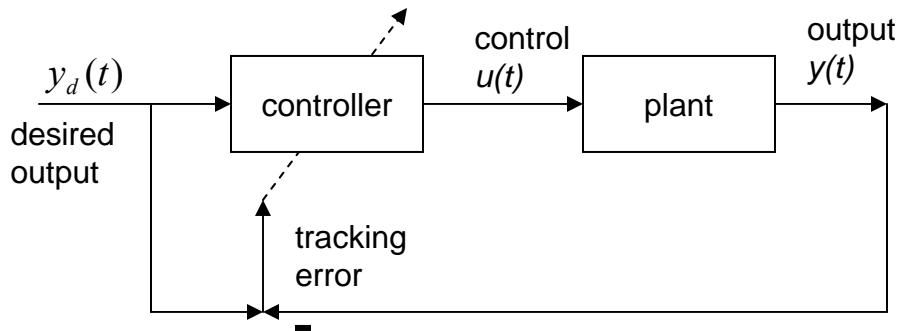
Controller Topologies



Indirect Scheme



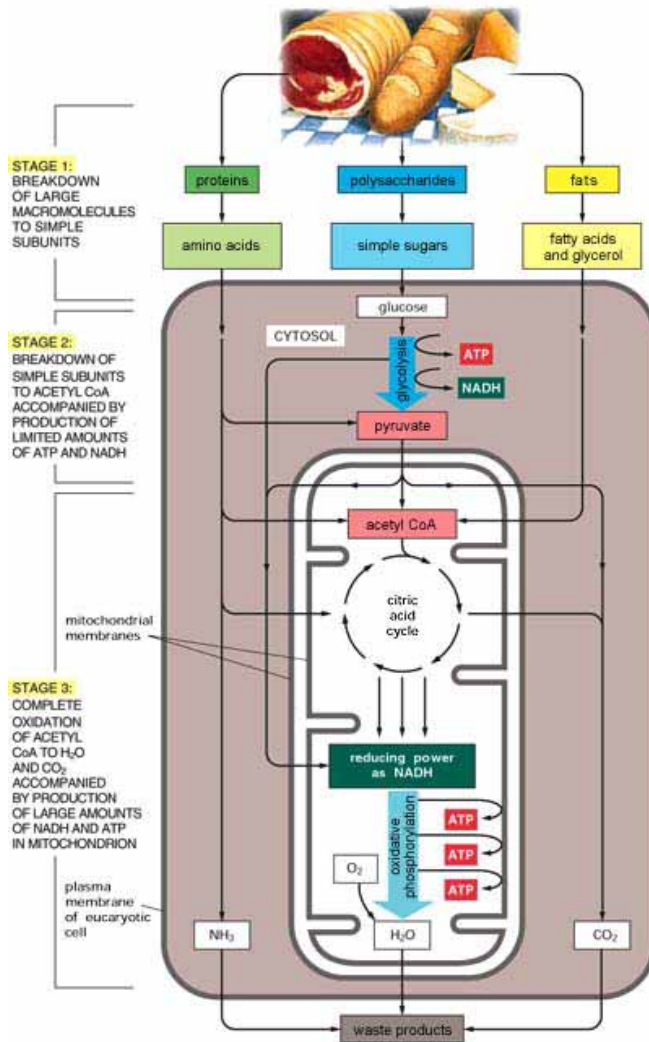
Feedback/Feedforward Scheme



Direct Scheme

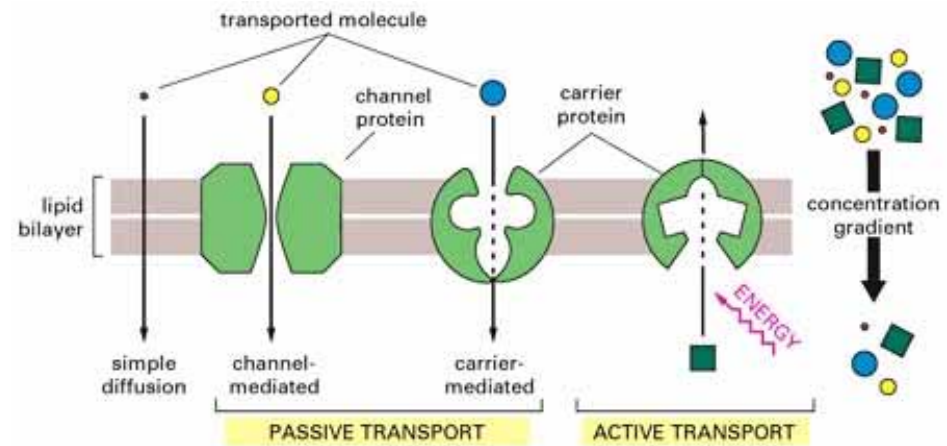
Optimality in Biological Systems

Cell Homeostasis



Cellular Metabolism

The individual cell is a complex feedback control system. It pumps ions across the cell membrane to maintain homeostasis, and has only **limited energy** to do so.



Permeability control of the cell membrane

<http://www.accessexcellence.org/RC/VL/GG/index.html>

Optimality in Control Systems Design

Rocket Orbit Injection

Dynamics

$$\dot{r} = w$$

$$\dot{w} = \frac{v^2}{r} - \frac{\mu}{r^2} + \frac{F}{m} \sin \phi$$

$$\dot{v} = \frac{-wv}{r} + \frac{F}{m} \cos \phi$$

$$\dot{m} = -Fm$$

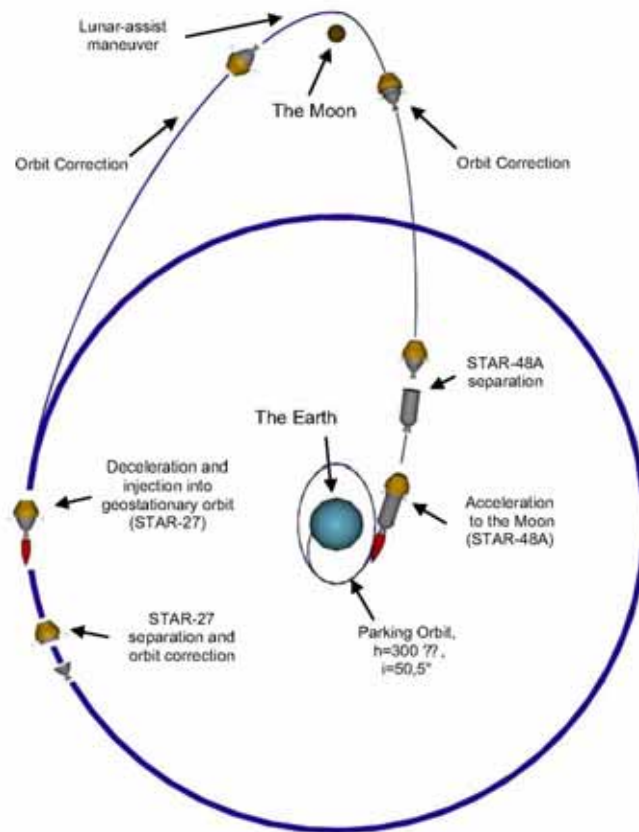


Fig. 1-1. Trajectory scheme

ISC Kosmotras Proprietary

Objectives

Get to orbit in minimum time

Use minimum fuel

Performance Index, Cost, or Value function

CT
$$J = \int_0^T [Q(x) + R(u)] dt = \int_0^T r(x, u) dt$$

DT
$$J = \sum_{k=0}^N r(x_k, u_k)$$

Strategic utility

utility

Minimum energy
$$r(x, u) = x^T Qx + u^T Ru$$

Minimum fuel
$$r(x, u) = |u|$$

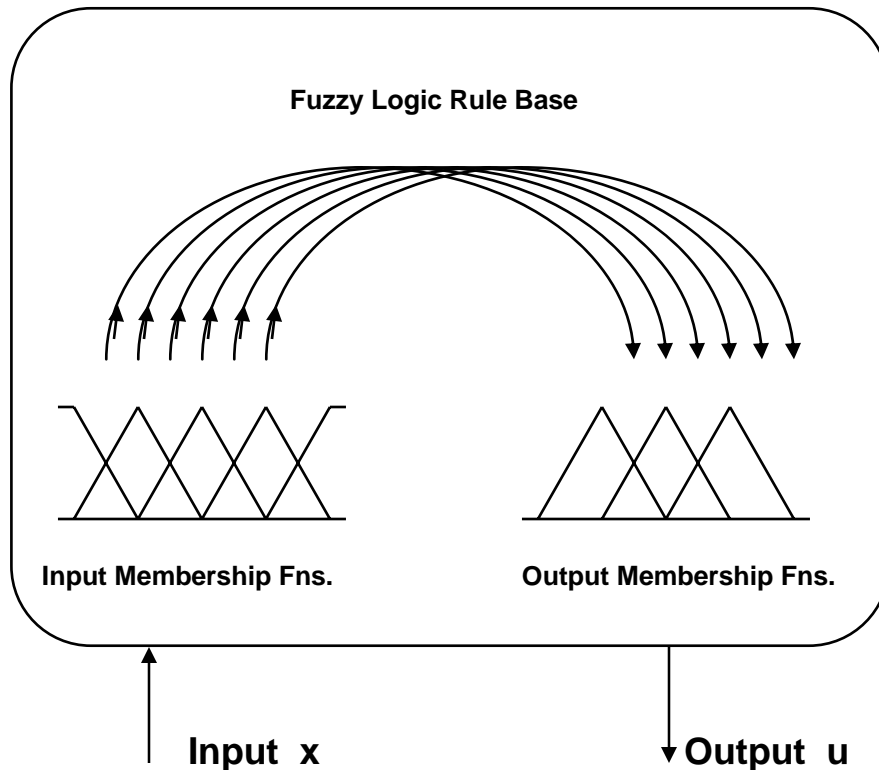
Minimum time
$$r(x, u) = 1$$
 Then
$$J = \int_0^T r(x, u) dt = T$$

Discounting
$$J = \sum_{k=0}^N \gamma^k r(x_k, u_k)$$

$$J = \int_0^T e^{-\rho t} r(x, u) dt$$

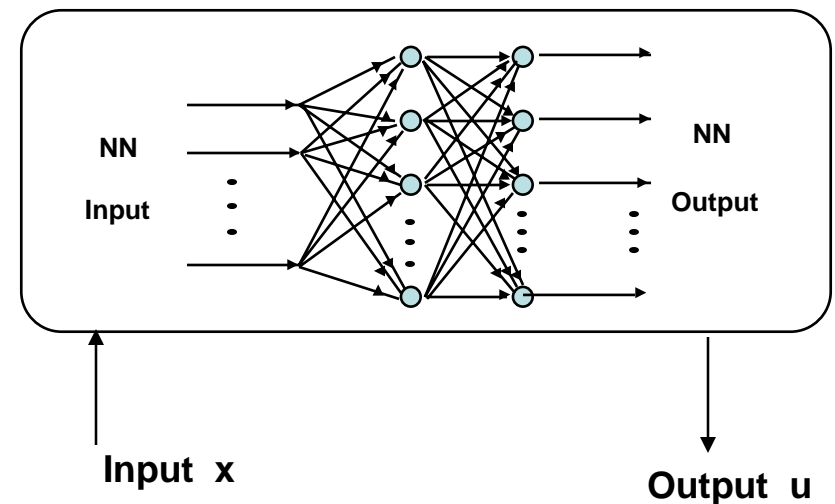
INTELLIGENT CONTROL TOOLS

Fuzzy Associative Memory (FAM)



Neural Network (NN)

(Includes Adaptive Control)



Both FAM and NN define a function $u = f(x)$ from inputs to outputs

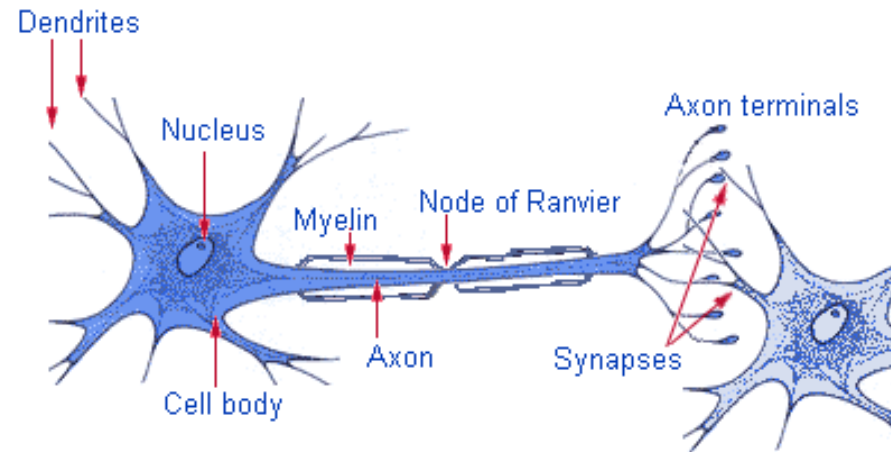
FAM and NN can both be used for:

- 1. Classification and Decision-Making**
- 2. Control**

NN Includes Adaptive Control (Adaptive control is a 1-layer NN)

Neural Network Properties

- Learning
- Recall
- Function approximation
- Generalization
- Classification
- Association
- Pattern recognition
- Clustering
- Robustness to single node failure
- Repair and reconfiguration



Nervous system cell.

<http://www.sirinet.net/~jgjohnso/index.html>

First groups working on NN Feedback Control in CS community

Werbos

Narendra

c. 1995

Sanner & Slotine

F.C. Chen & Khalil

Lewis

Polycarpou & Ioannou

Christodoulou & Rovithakis

A.J. Calise, McFarland, Naira Hovakimyan

Edgar Sanchez & Poznyak

Sam Ge, Zhang, et al.

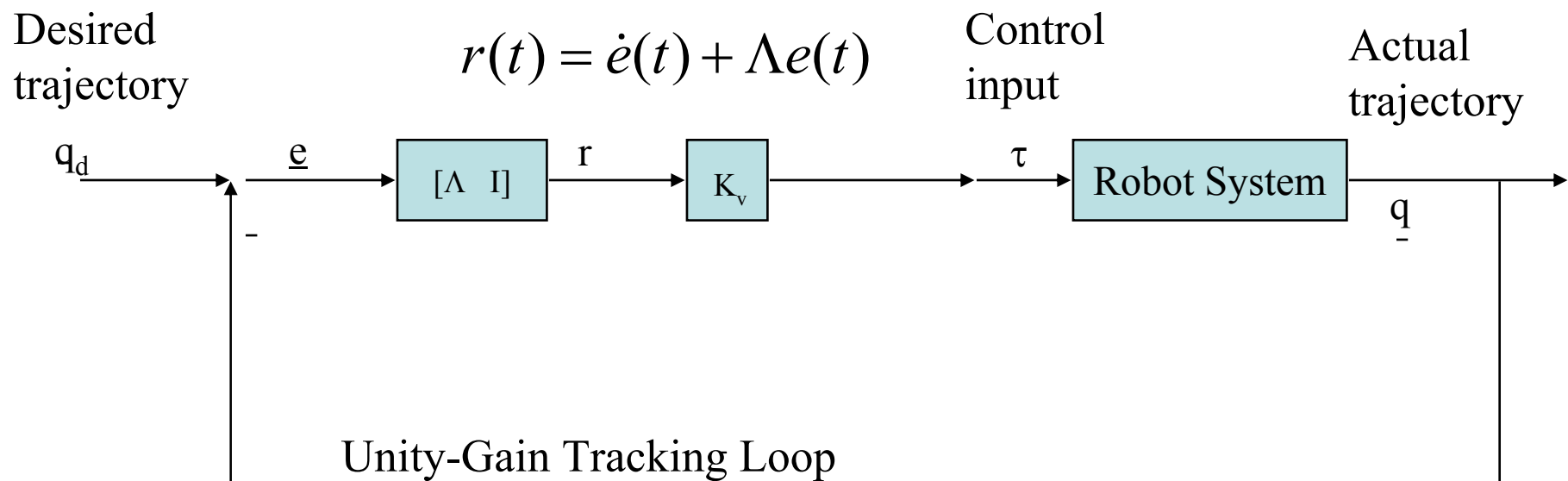
Jun Wang, Chinese Univ. Hong Kong

Industry Standard- PD Controller

Easy to implement with COTS controllers

Fast

Can be implemented with a few lines of code- e.g. MATLAB



But -- Cannot handle-

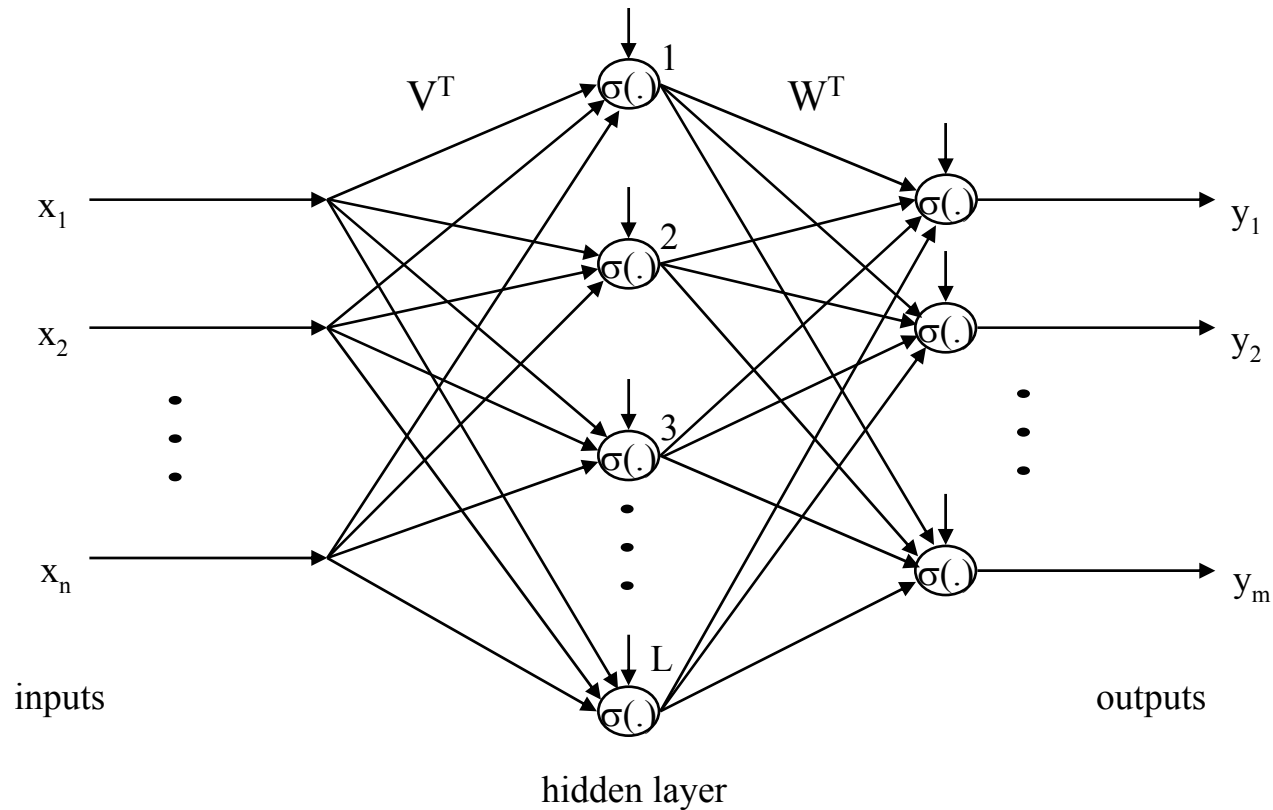
High-order unmodeled dynamics

Unknown disturbances

High performance specifications for nonlinear systems

Actuator problems such as friction, deadzones, backlash

Two-layer feedforward static neural network (NN)



Summation eqs

$$y_i = \sigma \left(\sum_{k=1}^K w_{ik} \sigma \left(\sum_{j=1}^n v_{kj} x_j + v_{k0} \right) + w_{i0} \right)$$

Matrix eqs

$$y = W^T \sigma(V^T x)$$

Control System Design Approach

Robot dynamics $M(q)\ddot{q} + V_m(q, \dot{q})\dot{q} + G(q) + F(\dot{q}) + \tau_d = \tau$

Tracking Error definition $e(t) = q_d(t) - q(t)$ $r = \dot{e} + \Lambda e$

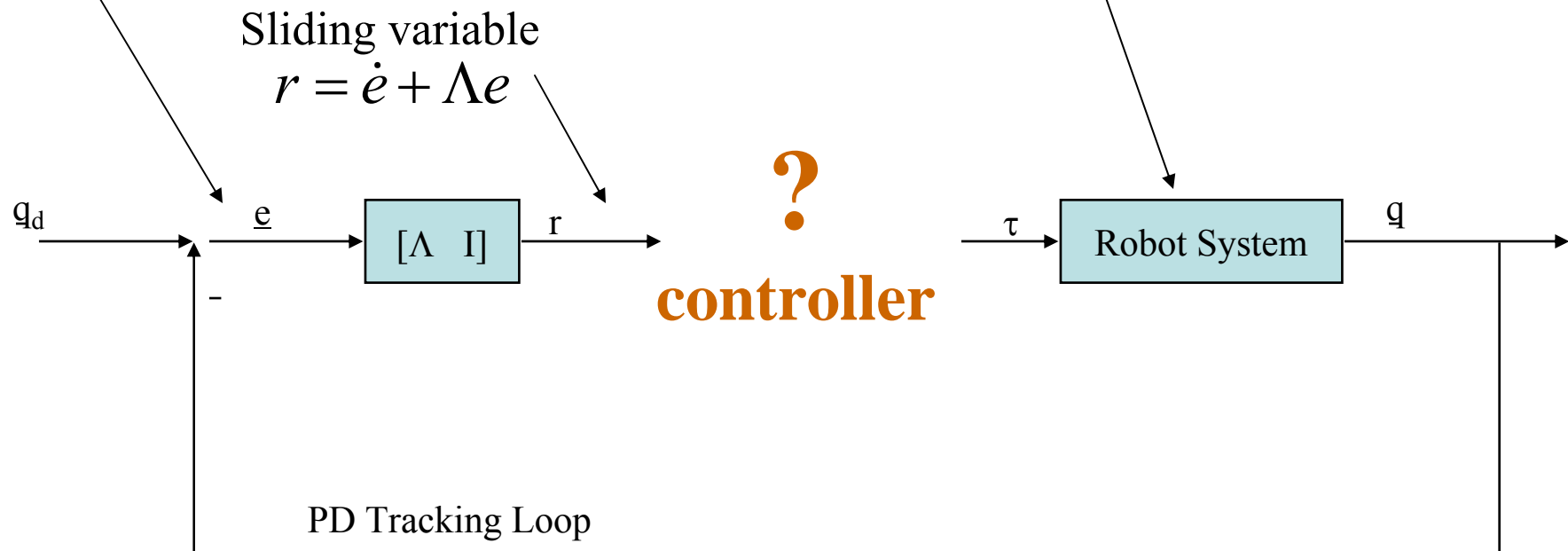
Error dynamics $M\dot{r} = -V_m r + f(x) + \tau_d - \tau$

Tracking error

$$e(t) = q_d(t) - q(t)$$

Robot dynamics

$$M(q)\ddot{q} + V_m(q, \dot{q})\dot{q} + G(q) + F(\dot{q}) + \tau_d = \tau$$



The equations give the FB controller structure

Control System Design Approach

Robot dynamics $M(q)\ddot{q} + V_m(q, \dot{q})\dot{q} + G(q) + F(\dot{q}) + \tau_d = \tau$

Tracking Error definition $e(t) = q_d(t) - q(t)$ $r = \dot{e} + \Lambda e$

Error dynamics $M\dot{r} = -V_m r + f(x) + \tau_d - \tau$

UNKNOWN FN.



Universal Approximation Property

Approx. unknown function by NN $f(x) = W^T \sigma(V^T x) + \varepsilon$

Define control input $\tau = \hat{W}^T \sigma(\hat{V}^T x) + K_v r - v$

Closed-loop dynamics

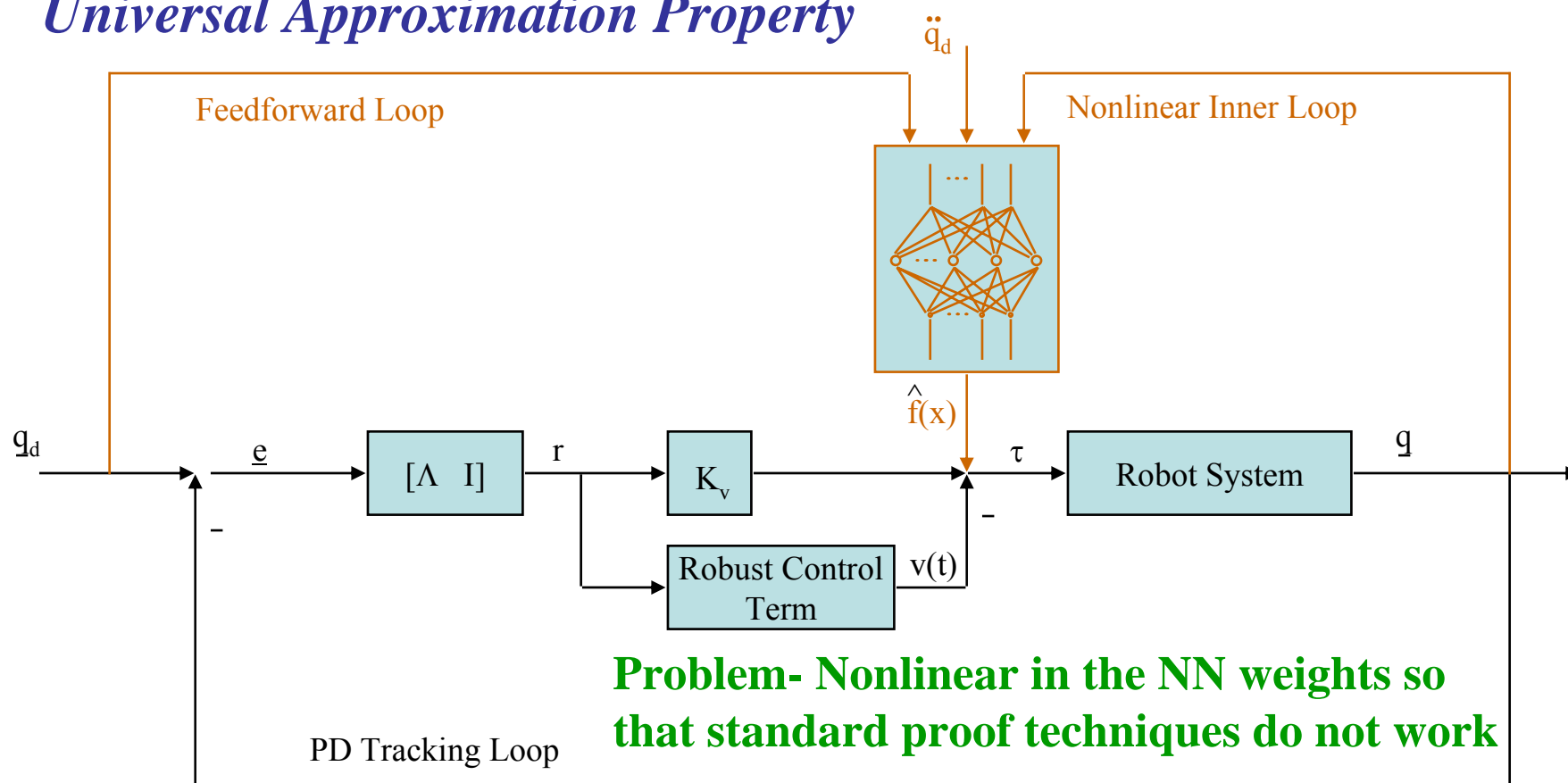
$$M\dot{r} = -V_m r - K_v r + W^T \sigma(V^T x) + \varepsilon - \hat{W}^T \sigma(\hat{V}^T x) + \tau_d + v(t)$$

$$M\dot{r} = -V_m r - K_v r + \tilde{f} + \tau_d + v(t)$$

Neural Network Robot Controller

Universal Approximation Property

Feedback linearization



Easy to implement with a few more lines of code

Learning feature allows for on-line updates to NN memory as dynamics change

Handles unmodelled dynamics, disturbances, actuator problems such as friction

NN universal basis property means no regression matrix is needed

Nonlinear controller allows faster & more precise motion

Stability Proof based on Lyapunov Extension

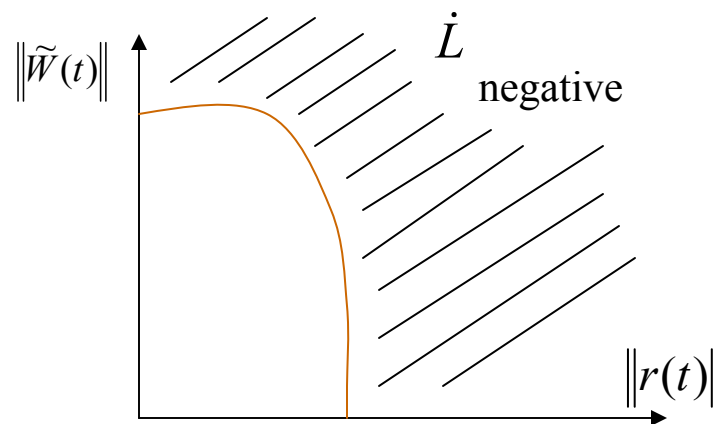
Define a Lyapunov Energy Function

$$L = \frac{1}{2} r^T M r + \frac{1}{2} \text{tr}(\tilde{W}^T \tilde{W}) + \frac{1}{2} \text{tr}(\tilde{V}^T \tilde{V})$$

Differentiate

$$\begin{aligned} \dot{L} = & -r^T K_v r + \frac{1}{2} r^T (\dot{M} - 2V_m) r \\ & + \text{tr} \tilde{W}^T (\dot{\tilde{W}} + \hat{\sigma} r^T - \hat{\sigma}' \hat{V}^T x r^T) \\ & + \text{tr} \tilde{V}^T (\dot{\tilde{V}} + x r^T \hat{W}^T \hat{\sigma}') + r^T (w + v) \end{aligned}$$

Using certain special tuning rules, one can show that the energy derivative is negative outside a compact set.



This proves that all signals are bounded

Problems—

1. How to characterize the NN weight errors as ‘small’?- use Frobenius Norm
2. Nonlinearity in the parameters requires extra care in the proof

Theorem 1 (NN Weight Tuning for Stability)

Let the desired trajectory $q_d(t)$ and its derivatives be bounded. Let the initial tracking error be within a certain allowable set U . Let Z_M be a known upper bound on the Frobenius norm of the unknown ideal weights Z .

Take the control input as

Can also use simplified tuning- Hebbian

$$\tau = \hat{W}^T \sigma(\hat{V}^T x) + K_v r - v \quad \text{with} \quad v(t) = -K_z (\|Z\|_F + Z_M) r.$$

Let weight tuning be provided by

Forward Prop term?

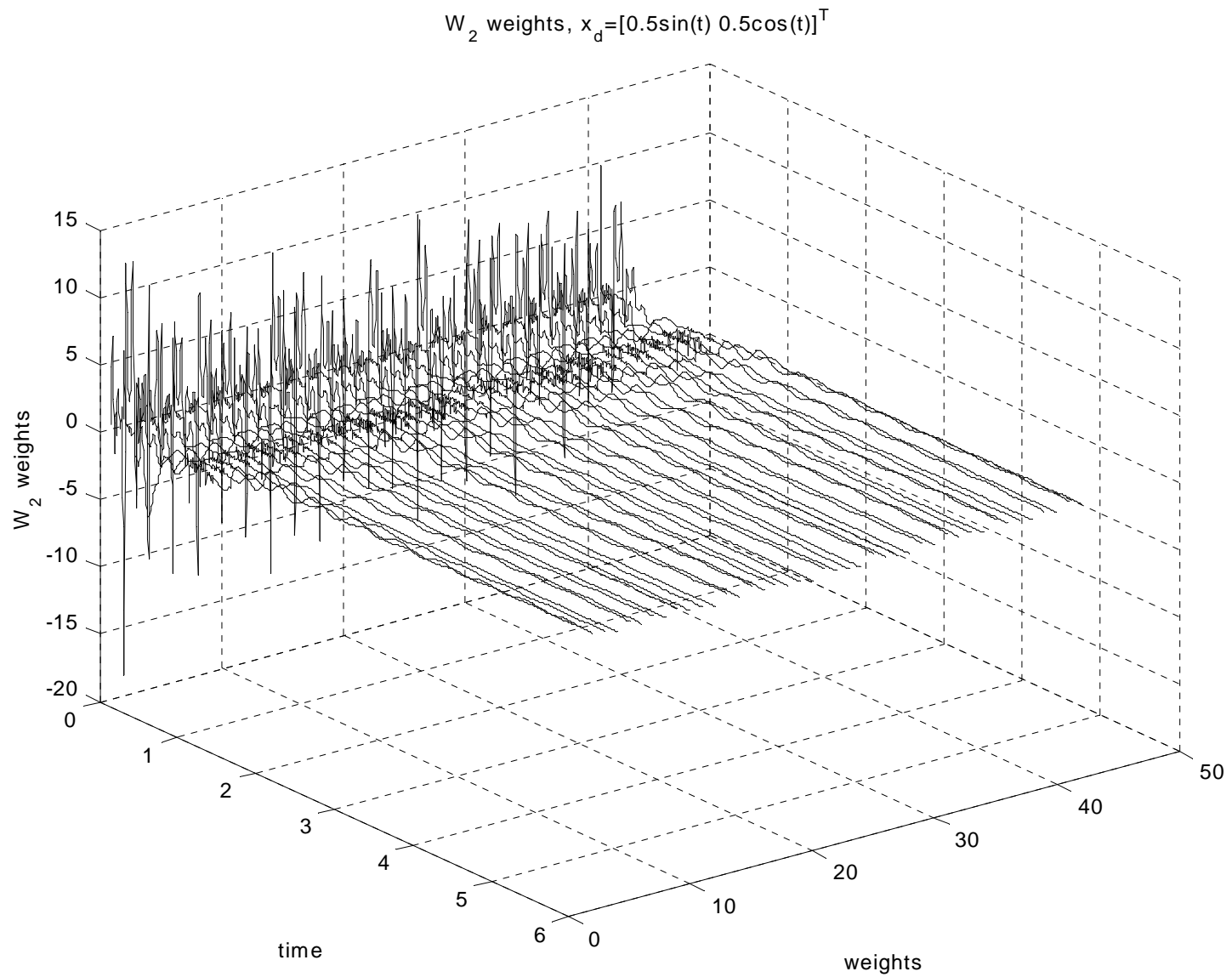
$$\dot{\hat{W}} = F \hat{\sigma} r^T - F \hat{\sigma}' \hat{V}^T x r^T - \kappa F \|r\| \hat{W}, \quad \dot{\hat{V}} = G x (\hat{\sigma}'^T \hat{W} r)^T - \kappa G \|r\| \hat{V}$$

with any constant matrices $F = F^T > 0, G = G^T > 0$, and scalar tuning parameter $\kappa > 0$. Initialize the weight estimates as $\hat{W} = 0, \hat{V} = random$.

Then the filtered tracking error $r(t)$ and NN weight estimates \hat{W}, \hat{V} are uniformly ultimately bounded. Moreover, arbitrarily small tracking error may be achieved by selecting large control gains K_v .

Backprop terms-
Werbos

Extra robustifying terms-
Narendra's e-mod extended to NLIP systems

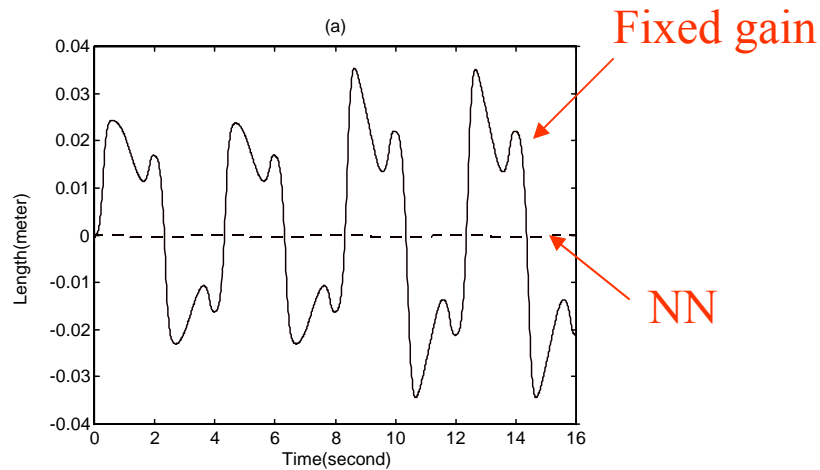
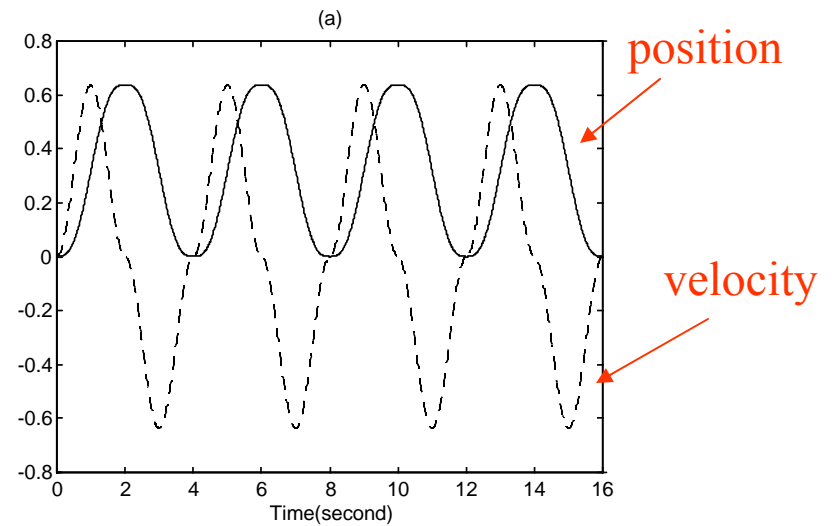


NN weights converge to the best learned values for the given system

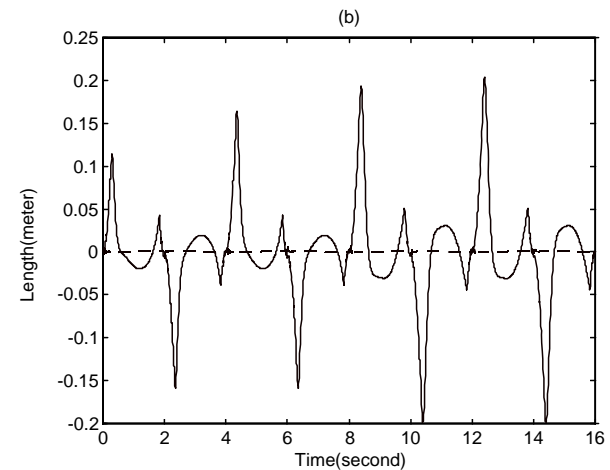
NN Friction Compensator

Trajectory Tracking Controller

Desired trajectory



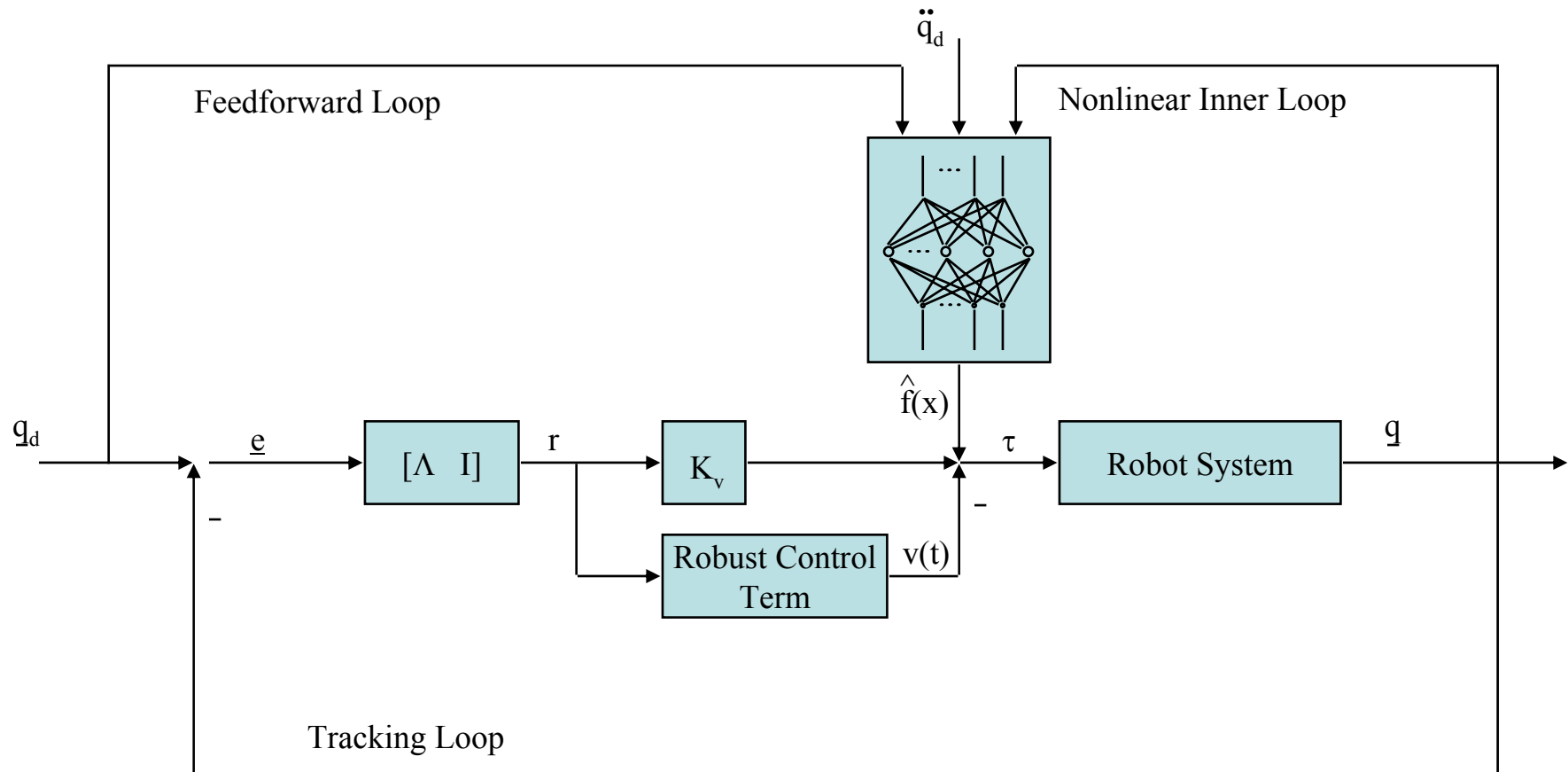
Position



Velocity

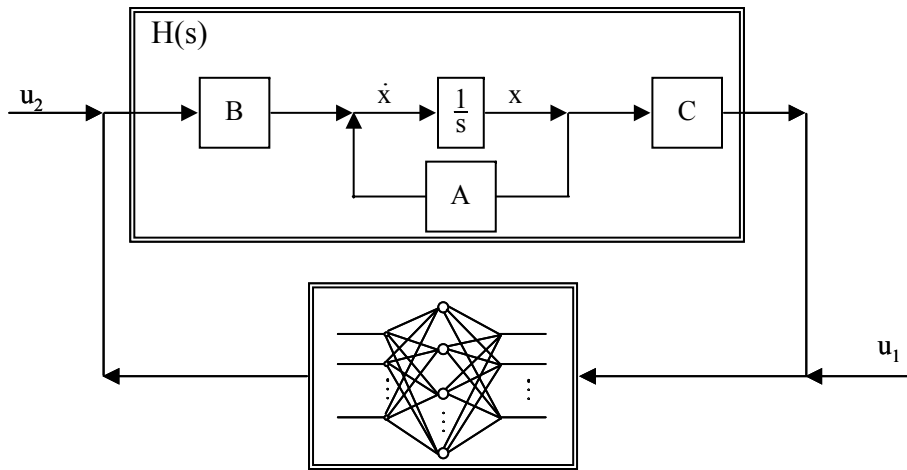
Tracking errors- solid = fixed gain controller, dashed= NN controller

Dynamic NN and Passivity



Static NN \Rightarrow Dynamic NN Feedback Controller

Closed-Loop System wrt Neural Network is a Dynamic (Recursive NN)



Discrete time case

$$x_{k+1} = Ax_k + W^T \sigma(V^T x_k) + u_k$$

The backprop tuning algorithms

$$\dot{\hat{W}} = F \hat{\sigma} r^T - F \hat{\sigma}' \hat{V}^T x r^T$$

$$\dot{\hat{V}} = Gx(\hat{\sigma}'^T \hat{W} r)^T$$

make the closed-loop system passive

The enhanced tuning algorithms

$$\dot{\hat{W}} = F \hat{\sigma} r^T - F \hat{\sigma}' \hat{V}^T x r^T - \kappa F \|r\| \hat{W}$$

$$\dot{\hat{V}} = Gx(\hat{\sigma}'^T \hat{W} r)^T - \kappa G \|r\| \hat{V}$$

make the closed-loop system **state-strict** passive

SSP gives extra robustness properties to disturbances and HF dynamics



Force Control



Flexible pointing systems



Vehicle active suspension

SBIR Contracts

What about practical Systems?

Flexible Systems with Vibratory Modes

Rigid dynamics

$$\begin{bmatrix} M_{rr} & M_{rf} \\ M_{fr} & M_{ff} \end{bmatrix} \begin{bmatrix} \ddot{q}_r \\ \ddot{q}_f \end{bmatrix} + \begin{bmatrix} V_{rr} & V_{rf} \\ V_{fr} & V_{ff} \end{bmatrix} \begin{bmatrix} \dot{q}_r \\ \dot{q}_f \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & K_{ff} \end{bmatrix} \begin{bmatrix} q_r \\ q_f \end{bmatrix} + \begin{bmatrix} F_r \\ 0 \end{bmatrix} + \begin{bmatrix} G_r \\ 0 \end{bmatrix} = \begin{bmatrix} B_r \\ B_f \end{bmatrix} \tau$$

Flexible dynamics

Problem- only one control input !

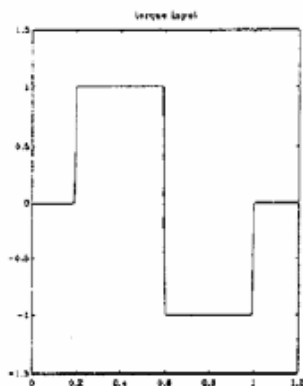
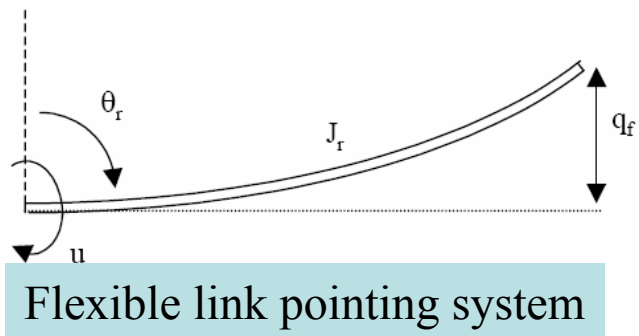


Fig. 2 Acceleration/deceleration torque profile.

acceleration

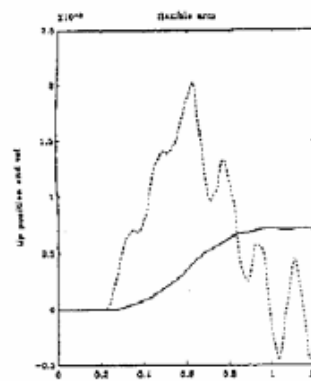


Fig. 3a Open-loop response of flexible arm. Tip position (solid) and vel. (dashed)

velocity
position

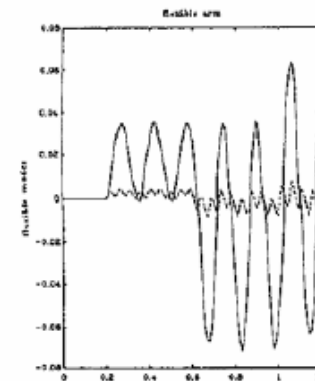
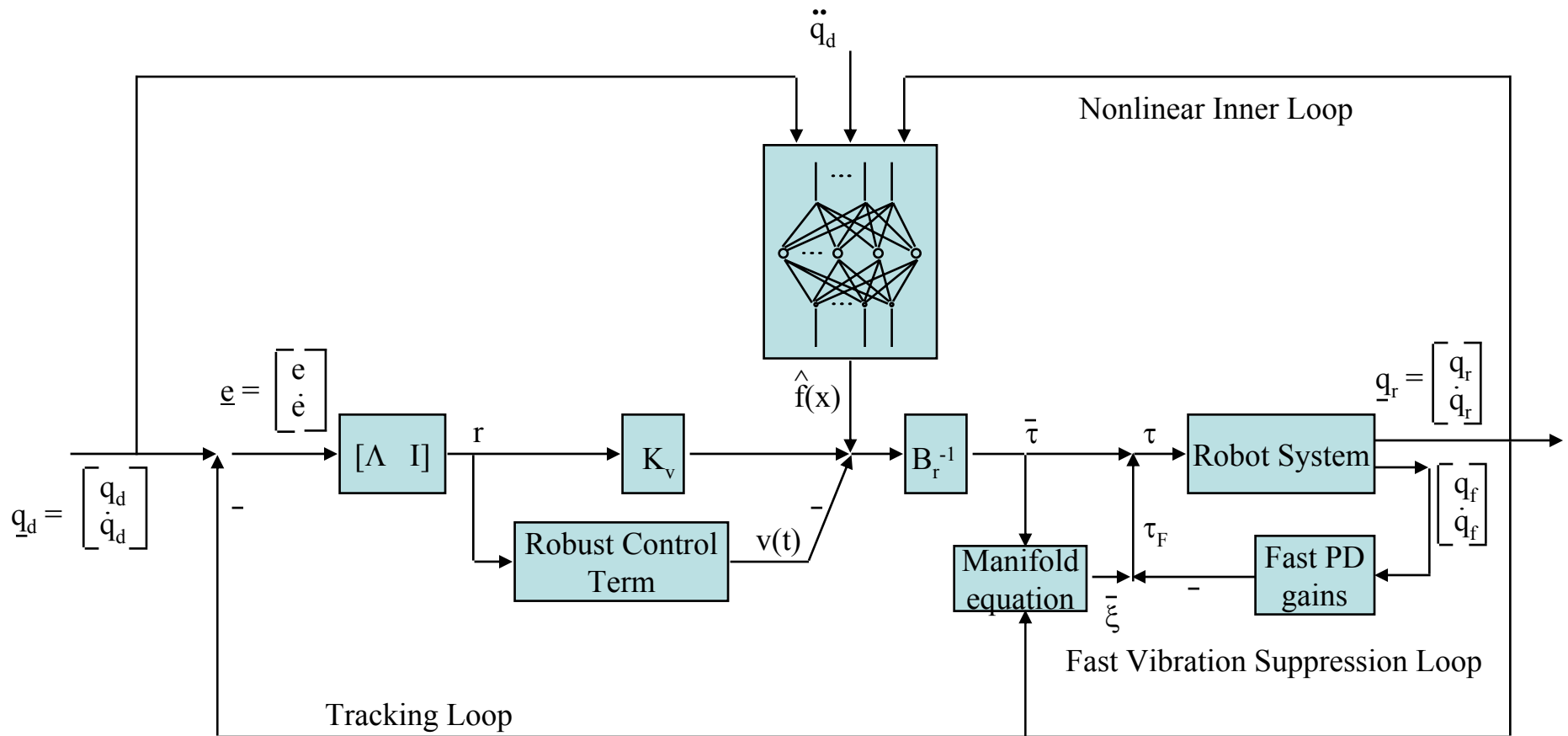


Fig. 3b Open-loop response of flexible arm. Flexible nodes.

Flex. modes

Singular Perturbations

Add an extra feedback loop
Use passivity to show stability



Neural network controller for Flexible-Link robot arm

Coupled Systems

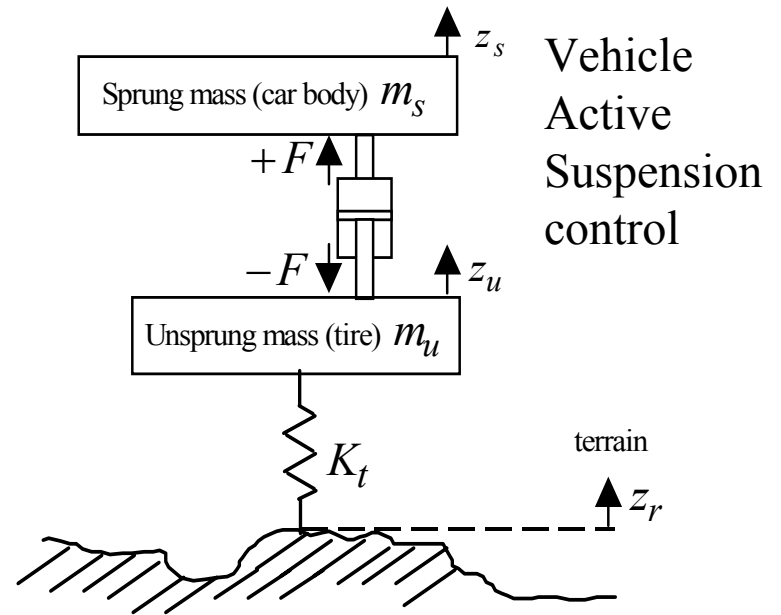
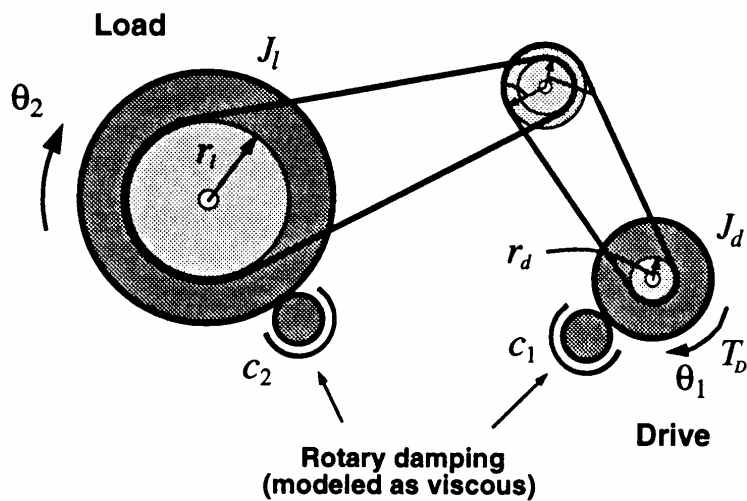
Problem- only one control input !

Robot mechanical dynamics

$$M(q)\ddot{q} + V_m(q, \dot{q})\dot{q} + F(\dot{q}) + G(q) + \tau_d = K_T i$$

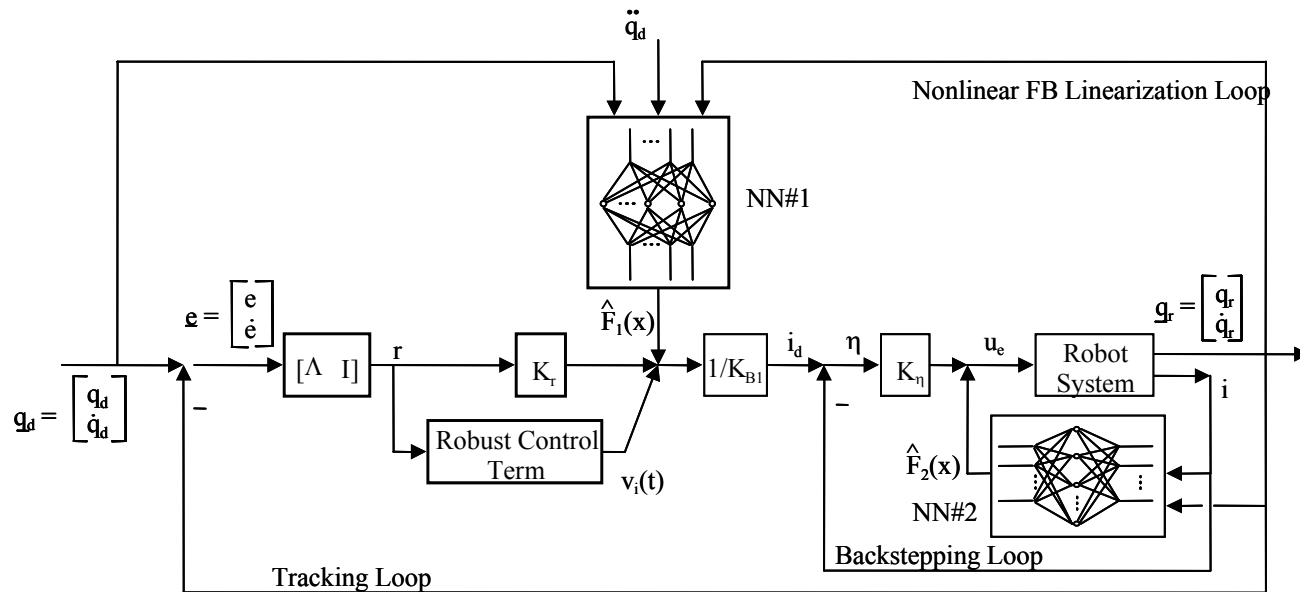
$$L\dot{i} + R(i, \dot{q}) + \tau_e = u_e$$

Motor electrical dynamics



Backstepping

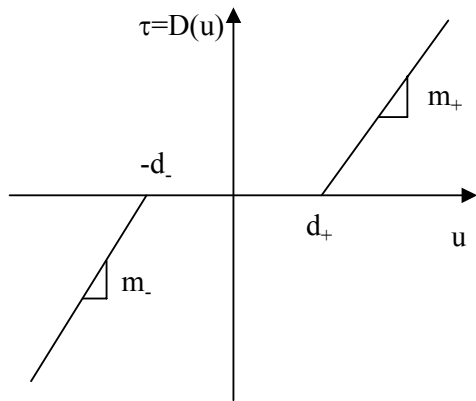
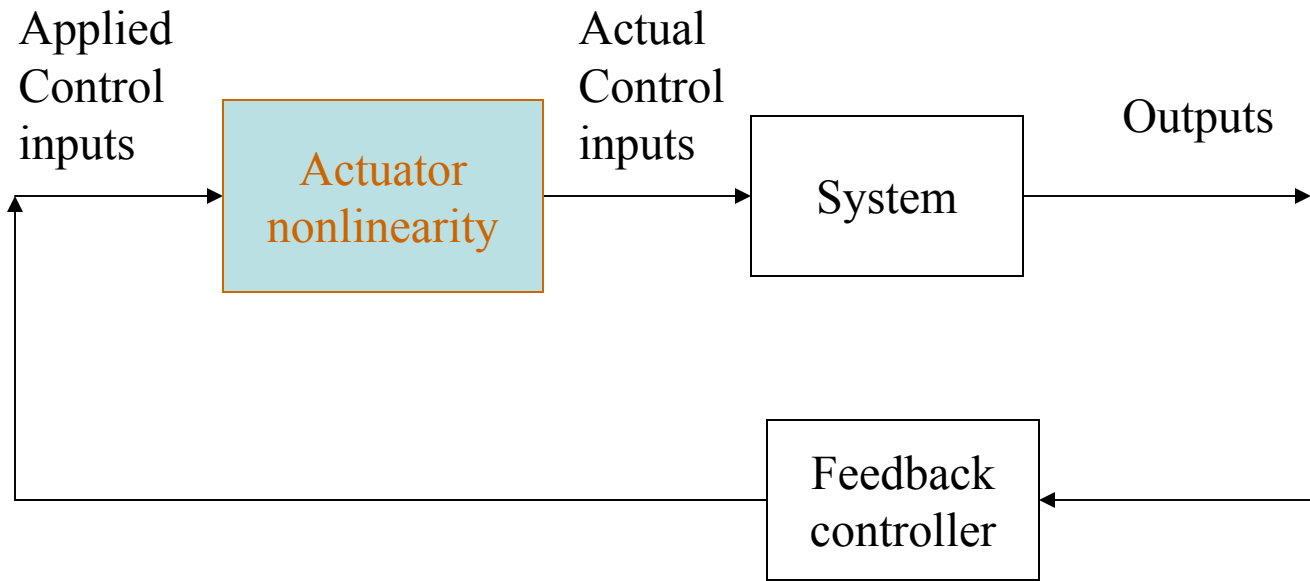
Add an extra feedback loop
Two NN needed
Use passivity to show stability



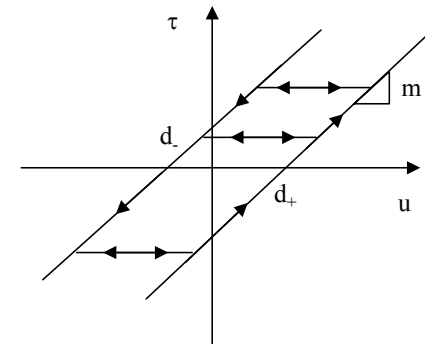
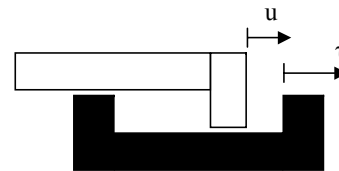
Neural network backstepping controller for Flexible-Joint robot arm

Advantages over traditional Backstepping- no regression functions needed

Actuator Nonlinearities

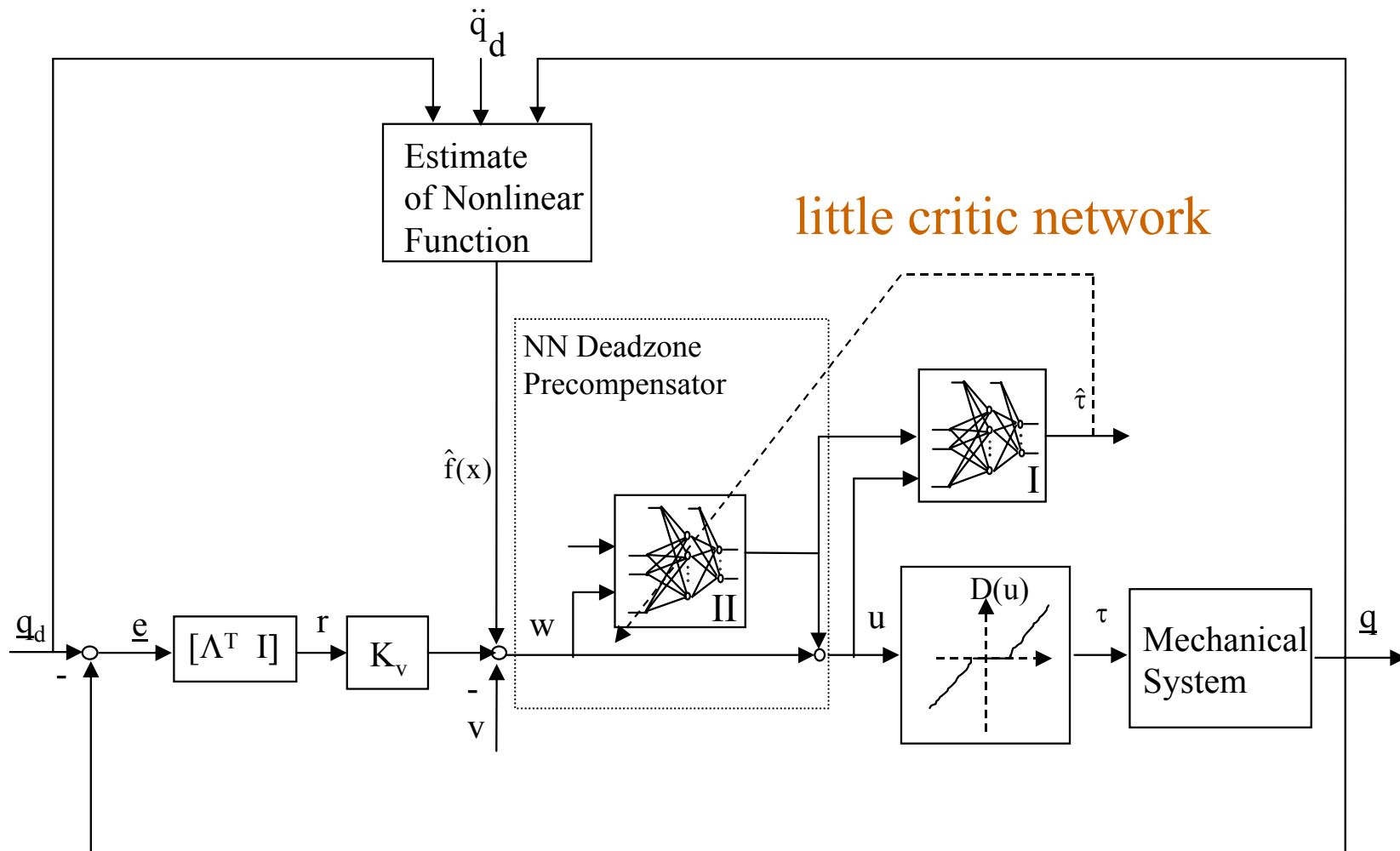


Deadzone



Backlash

NN in Feedforward Loop- Deadzone Compensation

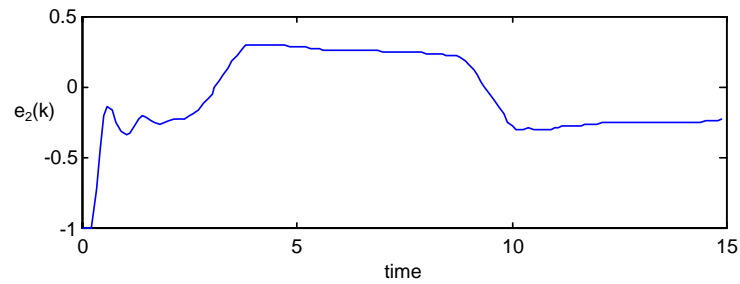
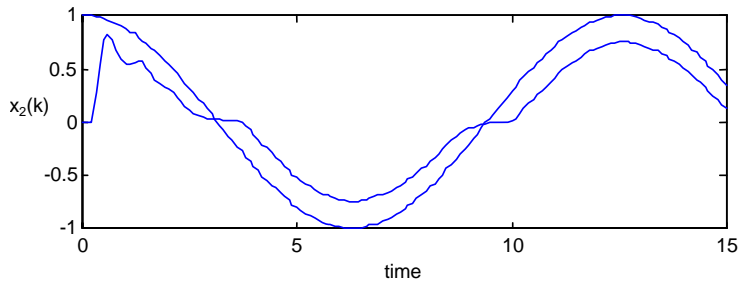


$$\hat{W}_i = T \sigma_i(U_i^T w) r^T \hat{W}^T \sigma'(U^T u) U^T - k_1 T \|r\| \hat{W}_i - k_2 T \|r\| \|\hat{W}_i\| \hat{W}_i$$

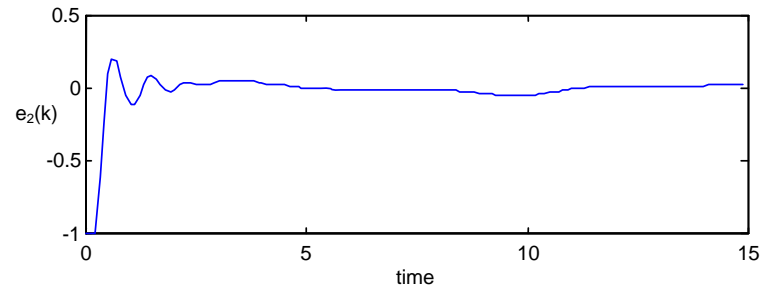
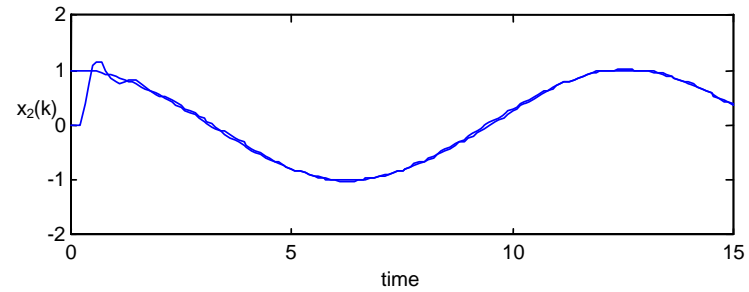
$$\hat{W} = -S \sigma'(U^T u) U^T \hat{W}_i \sigma_i(U_i^T w) r^T - k_1 S \|r\| \hat{W}$$

Acts like a 2-layer NN
With enhanced
backprop tuning !

Performance Results

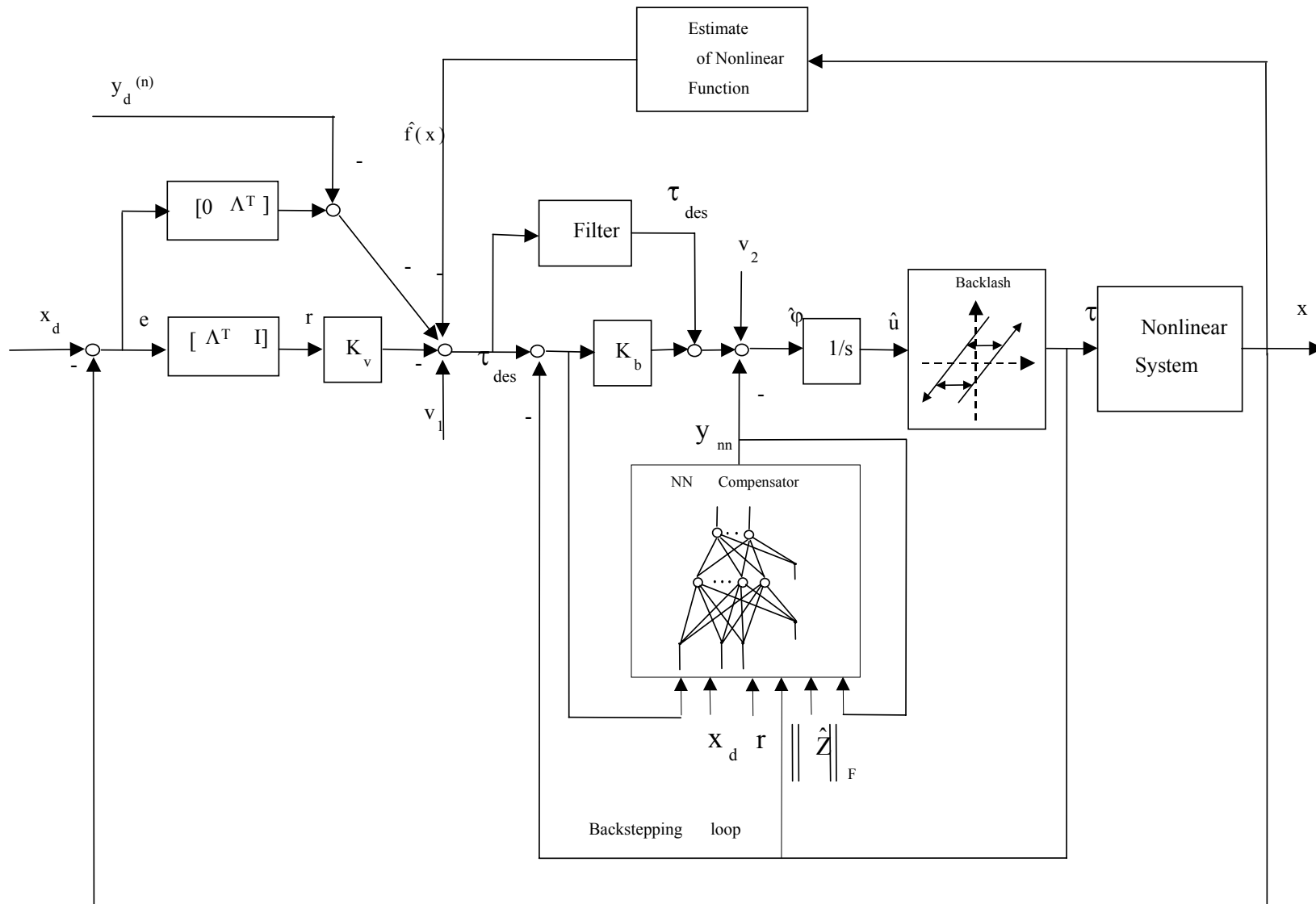


PD control-
deadzone chops out the middle



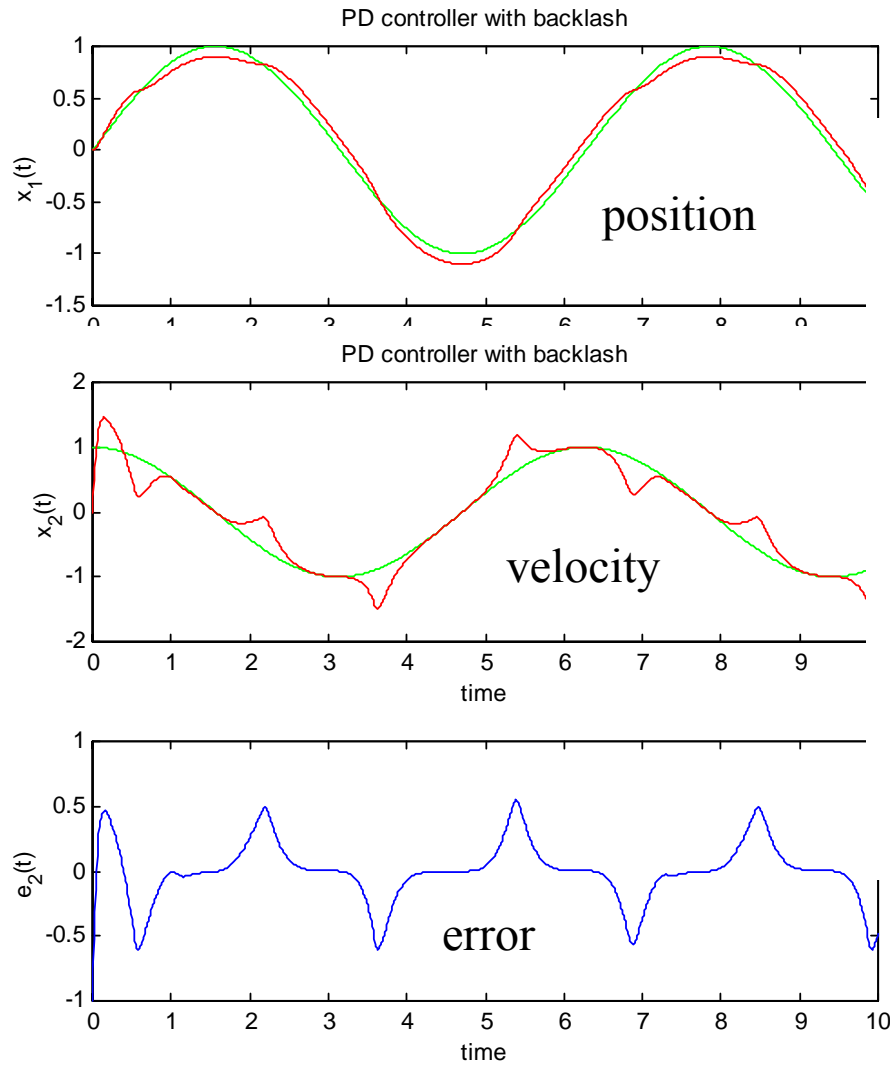
NN control fixes the problem

Dynamic inversion NN compensator for system with Backlash

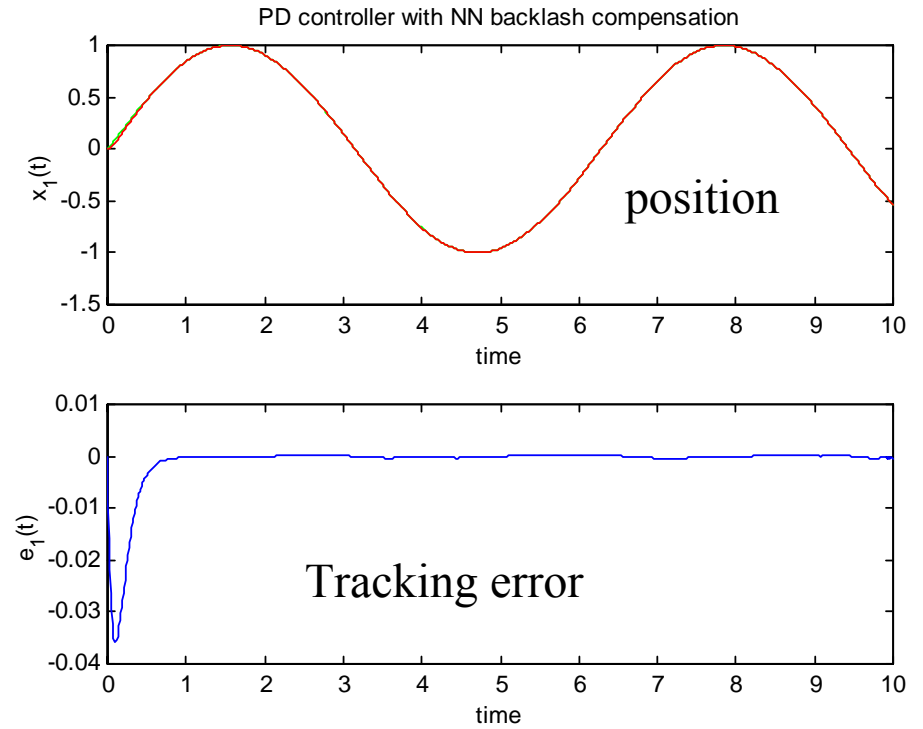


U.S. patent- Selmic, Lewis, Calise, McFarland

Performance Results



PD control-
backlash chops off tops & bottoms



NN control fixes the problem

NN Observers

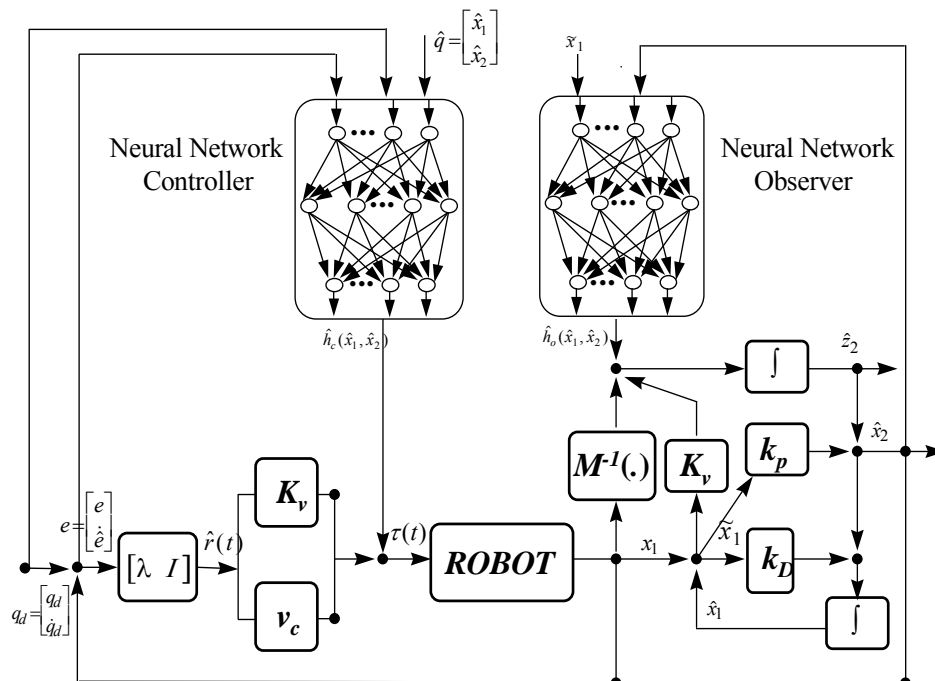
Needed when all states are not measured

$$\dot{\hat{\mathbf{z}}}_1 = \hat{\mathbf{x}}_2 + k_D \tilde{\mathbf{x}}_1$$

$$\dot{\hat{\mathbf{z}}}_2 = \hat{\mathbf{W}}_o^T \sigma_o(\hat{\mathbf{x}}_1, \hat{\mathbf{x}}_2) + \mathbf{M}^{-1}(\mathbf{x}_1) \tau(t) + \mathbf{K} \tilde{\mathbf{x}}_1$$

$$\hat{\mathbf{r}}(t) = \dot{\hat{\mathbf{e}}}(t) + \Lambda \mathbf{e}(t)$$

$$\tau(t) = \hat{\mathbf{W}}_c^T \sigma_c(\hat{\mathbf{x}}_1, \hat{\mathbf{x}}_2) + \mathbf{K}_v \hat{\mathbf{r}}(t) - \mathbf{v}_c(t)$$



$$\begin{aligned} \dot{\hat{\mathbf{W}}}_o &= -k_D \mathbf{F}_o \sigma_o(\hat{\mathbf{x}}) \tilde{\mathbf{x}}_1^T \\ &\quad - \kappa_o \mathbf{F}_o \|\tilde{\mathbf{x}}_1\| \hat{\mathbf{W}}_o - \kappa_o \mathbf{F}_o \hat{\mathbf{W}}_o \end{aligned}$$

$$\begin{aligned} \dot{\hat{\mathbf{W}}}_c &= \mathbf{F}_c \sigma_c(\hat{\mathbf{x}}_1, \hat{\mathbf{x}}_2) \hat{\mathbf{r}}^T \\ &\quad - \kappa_c \mathbf{F}_c \|\hat{\mathbf{r}}\| \hat{\mathbf{W}}_c \end{aligned}$$

NN Control for Discrete Time Systems

dynamics

$$x(k+1) = f(x(k)) + g(x(k))u(k)$$

Gradient descent with momentum

NN Tuning

$$\hat{W}_i(k+1) = \hat{W}_i(k) - \alpha_i \hat{\phi}_i(k) \hat{y}_i^T(k) - \Gamma \left\| I - \alpha_i \hat{\phi}_i(k) \hat{\phi}_i^T(k) \right\| \hat{W}_i(k)$$

Extra robust term

Error-based tuning

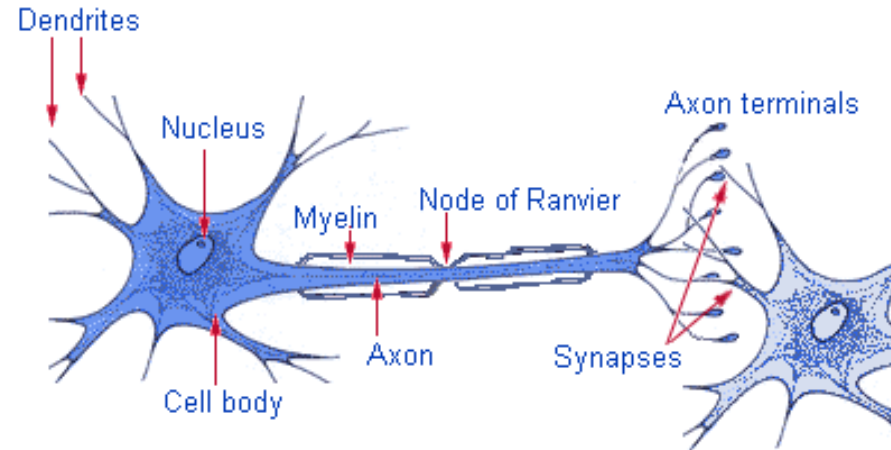
$$\hat{y}_i(k) \equiv \hat{W}_i^T(k) \hat{\phi}_i(k) + K_v r(k), \quad \text{for } i = 1, \dots, N-1 \quad \text{and} \quad \hat{y}_N(k) \equiv r(k+1), \quad \text{for last layer}$$

U.S. Patent- Jagannathan, Lewis

Neural Network Properties

USED

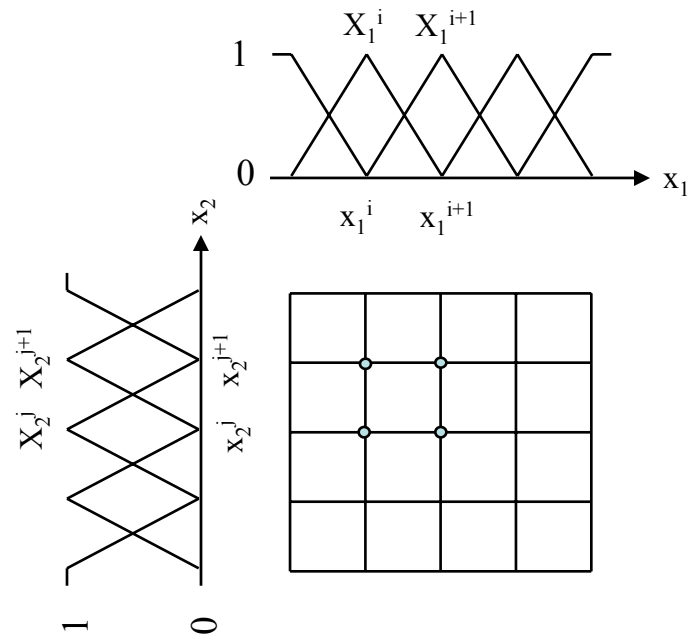
- Learning
- Recall
- Function approximation
- Generalization
- Classification
- Association
- Pattern recognition
- Clustering
- Robustness to single node failure ???
- Repair and reconfiguration



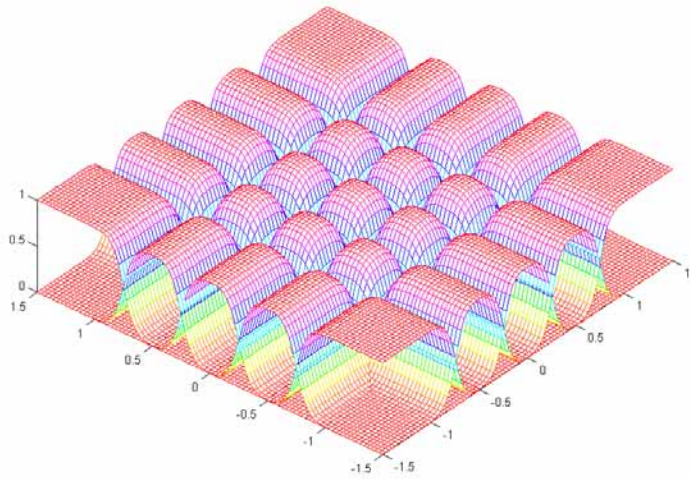
Nervous system cell.

<http://www.sirinet.net/~jgjohnso/index.html>

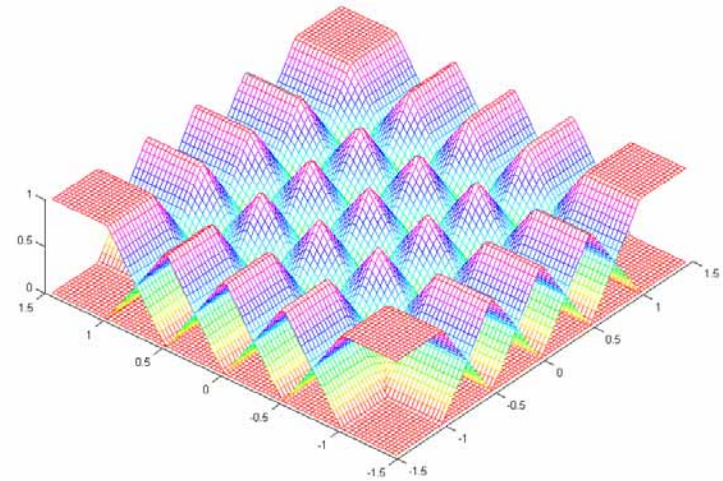
Relation Between Fuzzy Systems and Neural Networks



FL Membership Functions for 2-D Input Vector x

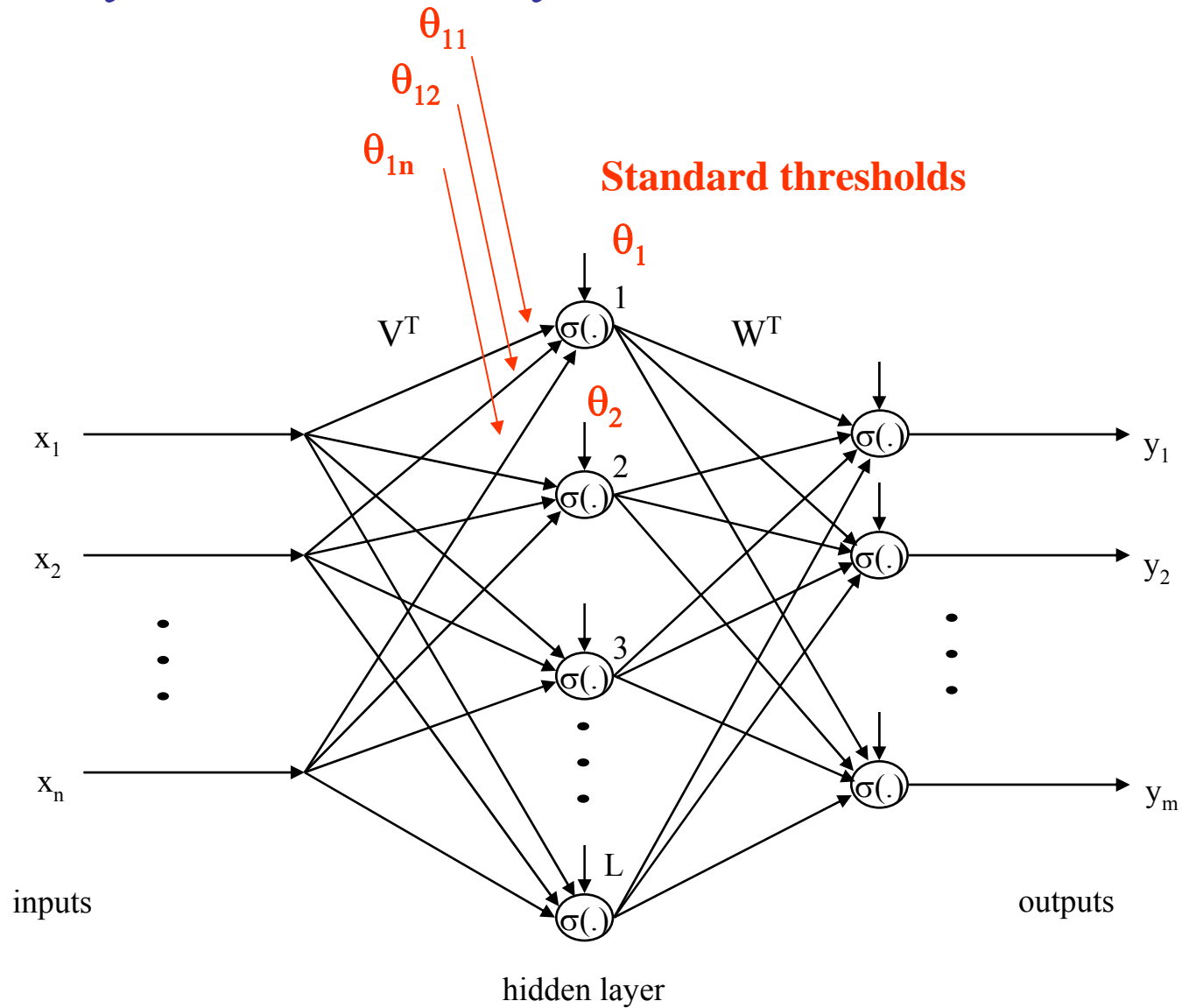


Separable Gaussian activation functions for RBF NN



Separable triangular activation functions for CMAC NN

Two-layer NN as FL System



FL system = NN with VECTOR thresholds

Fuzzy Logic Controllers

Gaussian membership function

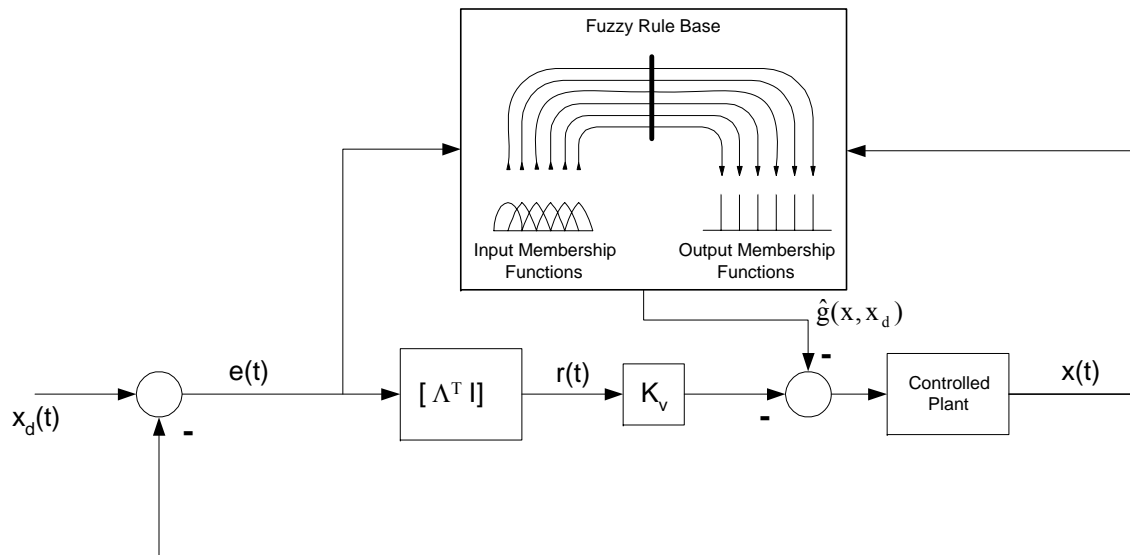
$$\phi_{A_i^l}(z_i, a_i^l, b_i^l) = e^{\left(-a_i^{l^2} (z_i - b_i^l)^2\right)}.$$

Tuning laws

$$\dot{\hat{a}} = K_a A^T \hat{W} r - k_a K_a \hat{a} \|r\|$$

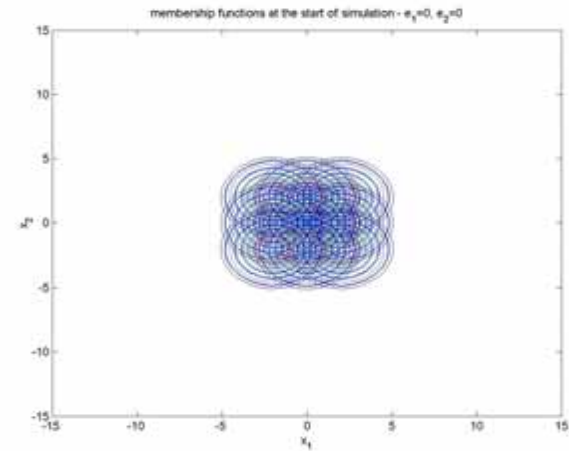
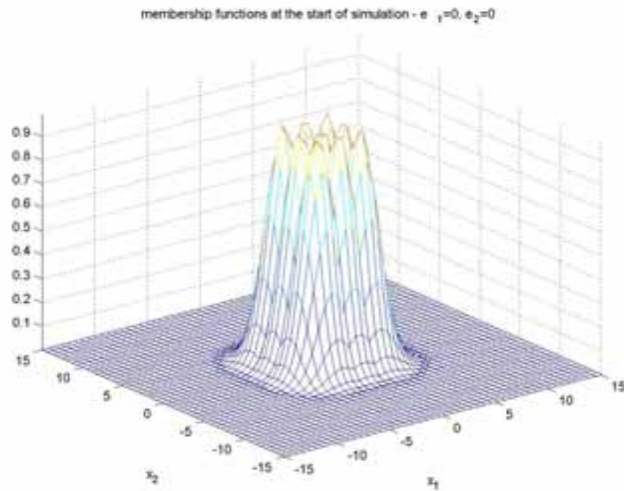
$$\dot{\hat{b}} = K_b B^T \hat{W} r - k_b K_b \hat{b} \|r\|$$

$$\dot{\hat{W}} = K_w (\hat{\Phi} - A\hat{a} - B\hat{b}) r^T - k_w K_w \hat{W} \|r\|$$

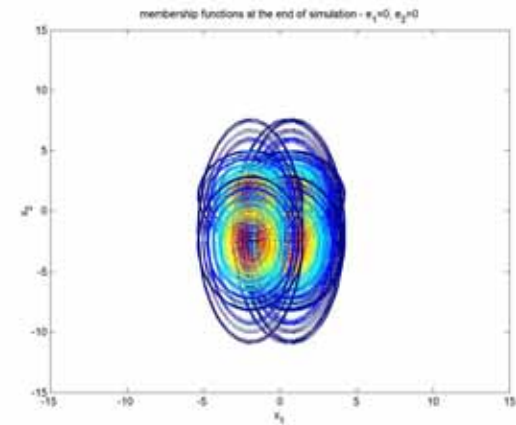
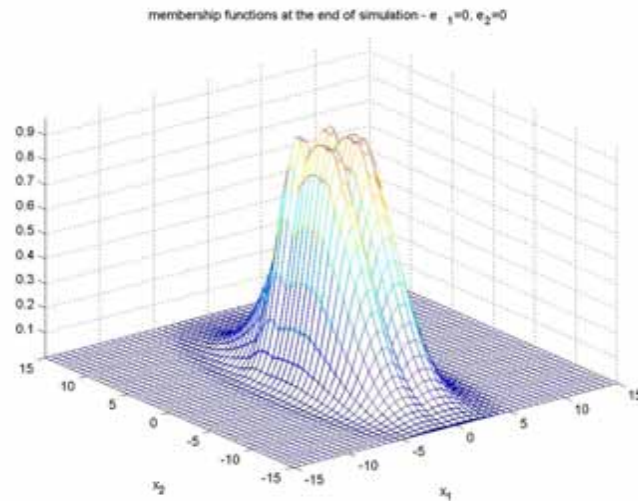


Dynamic Focusing of Awareness

Initial MFs



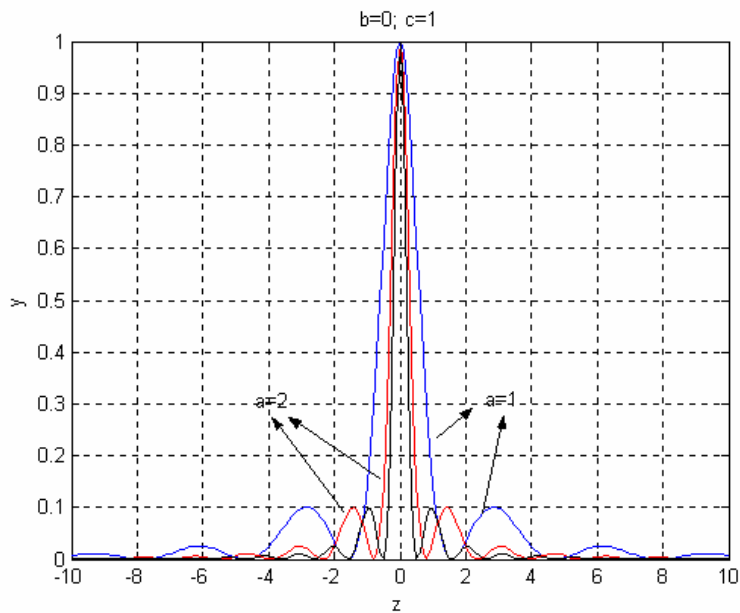
Final MFs



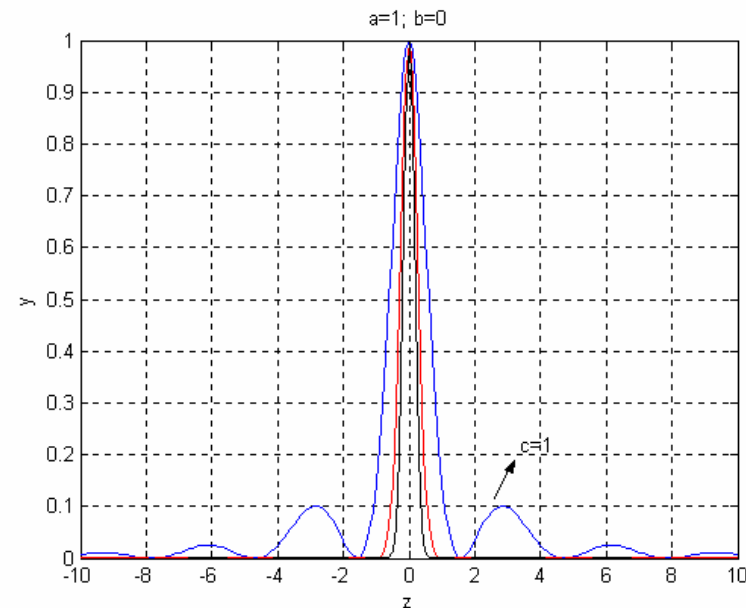
Elastic Fuzzy Logic- c.f. P. Werbos

$\phi(z, a, b, c) = \phi_B(z, a, b)^{c^2}$ ← Weights importance of factors in the rules

$$\phi(z, a, b, c) = \left[\frac{\cos^2(a(z-b))}{1 + a^2(z-b)^2} \right]^{c^2}$$



Effect of change of membership function spread "a"



Effect of change of membership function elasticities "c"

Elastic Fuzzy Logic Control

Control

$$u(t) = -K_v r - \hat{g}(x, x_d)$$

Tune Control Rep. Values

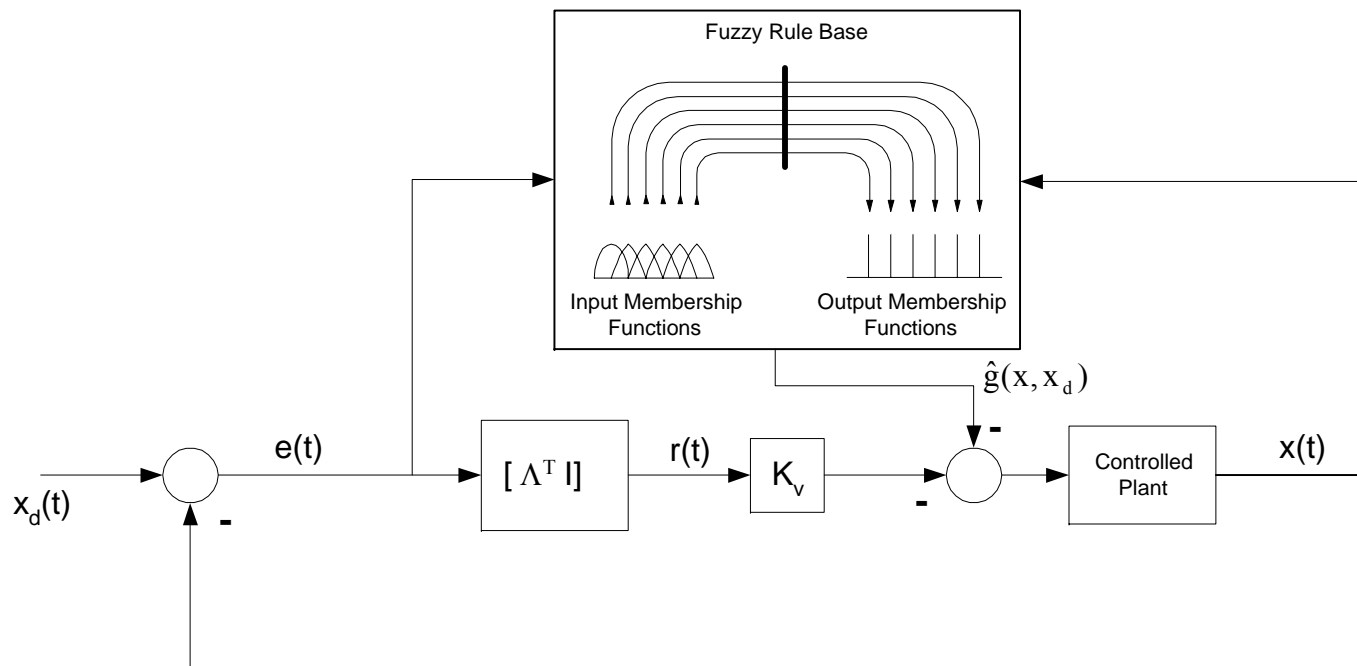
$$\dot{\hat{W}} = K_w (\hat{\Phi} - A\hat{a} - B\hat{b} - C\hat{c}) r^T - k_w K_w \hat{W} \|r\|$$

Tune Membership Functions

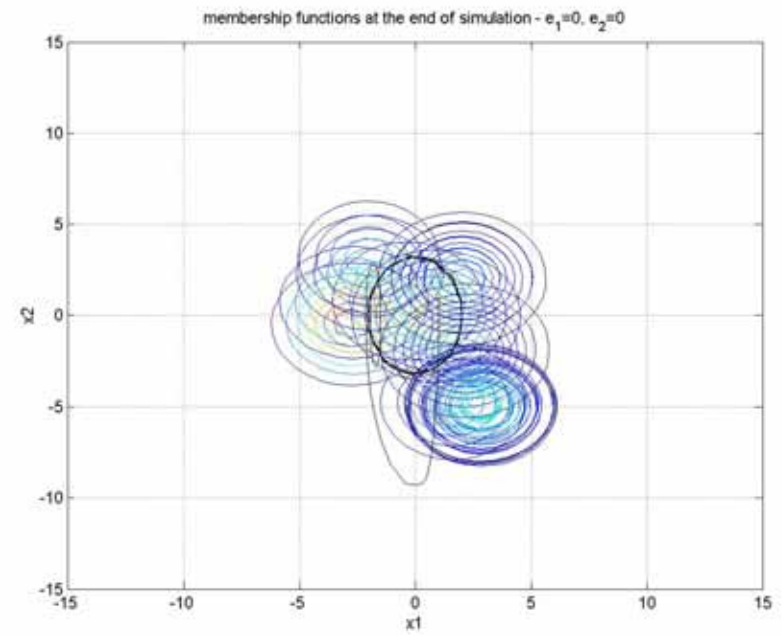
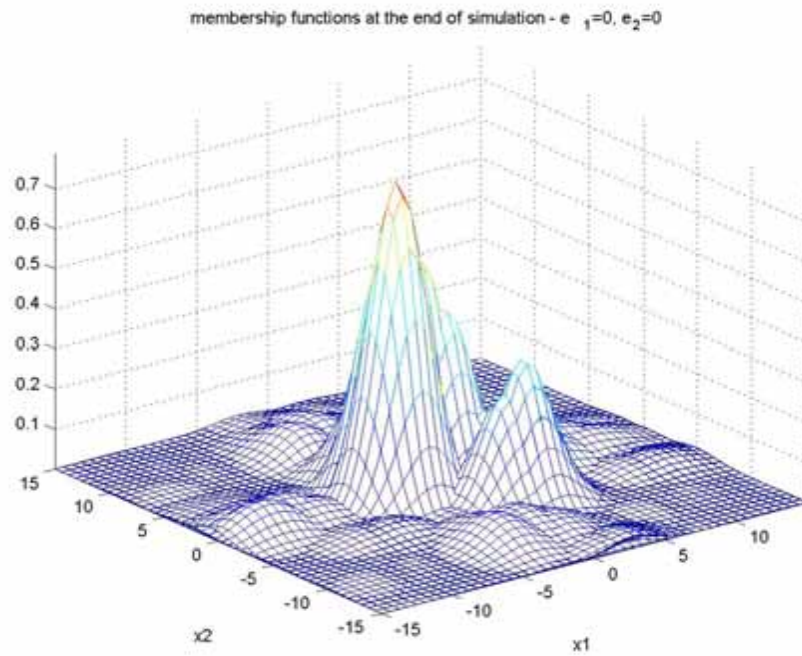
$$\dot{\hat{a}} = K_a A^T \hat{W} r - k_a K_a \hat{a} \|r\|$$

$$\dot{\hat{b}} = K_b B^T \hat{W} r - k_b K_b \hat{b} \|r\|$$

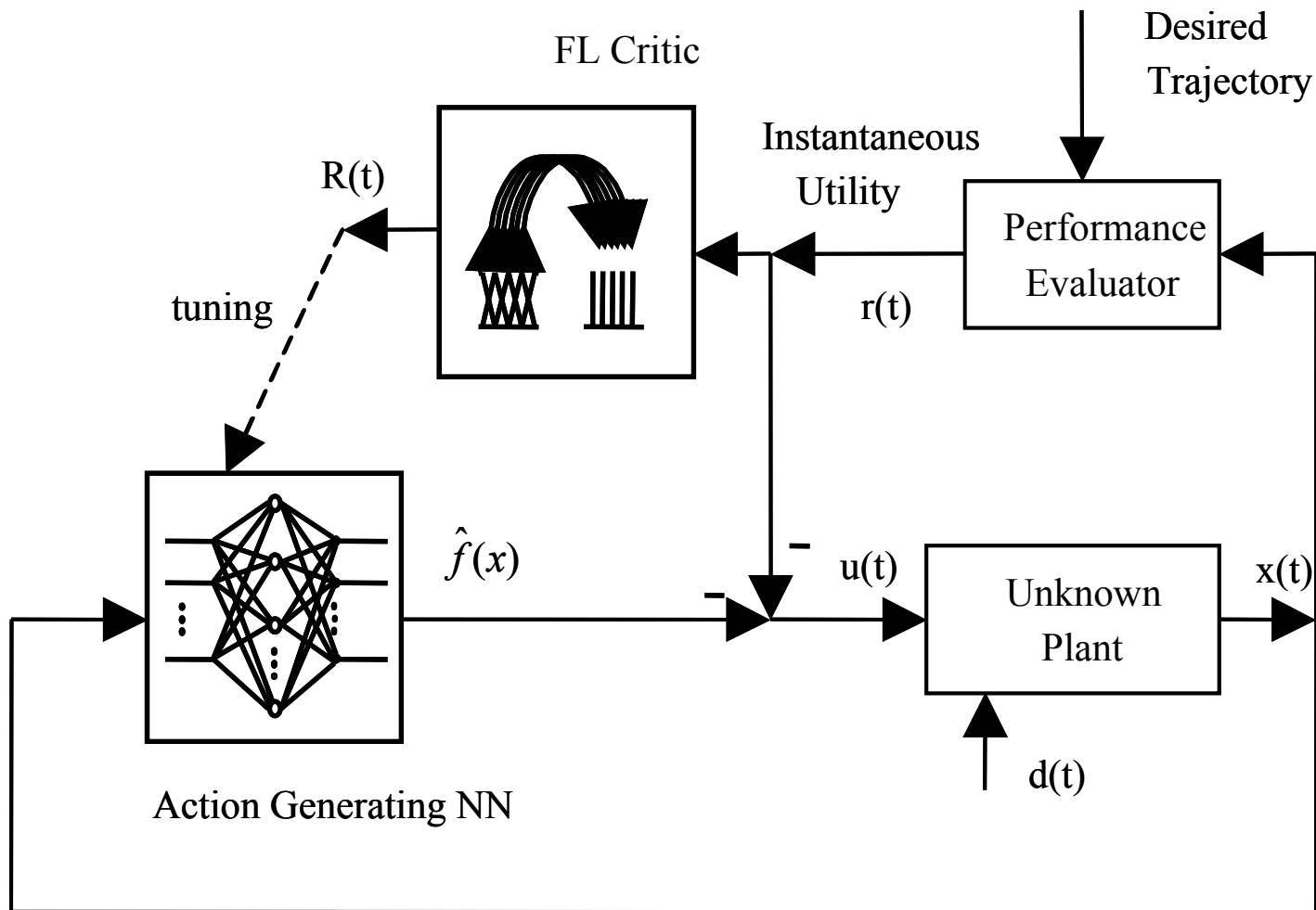
$$\dot{\hat{c}} = K_c C^T \hat{W} r - k_c K_c \hat{c} \|r\|$$



Better Performance



Fuzzy Logic Critic NN controller



Learning FL Critic Controller

Tune Action generating NN (controller)

$$\dot{\hat{W}}_2 = \Gamma \frac{\sigma(\chi_2)}{2} r^T - \Gamma \frac{\sigma(\chi_2)}{2} R^T \hat{W}_1^T \mu'(\hat{V}_1^T r) \hat{V}_1^T - \Gamma \frac{\hat{W}_2}{2}$$

Tune Fuzzy Logic Critic

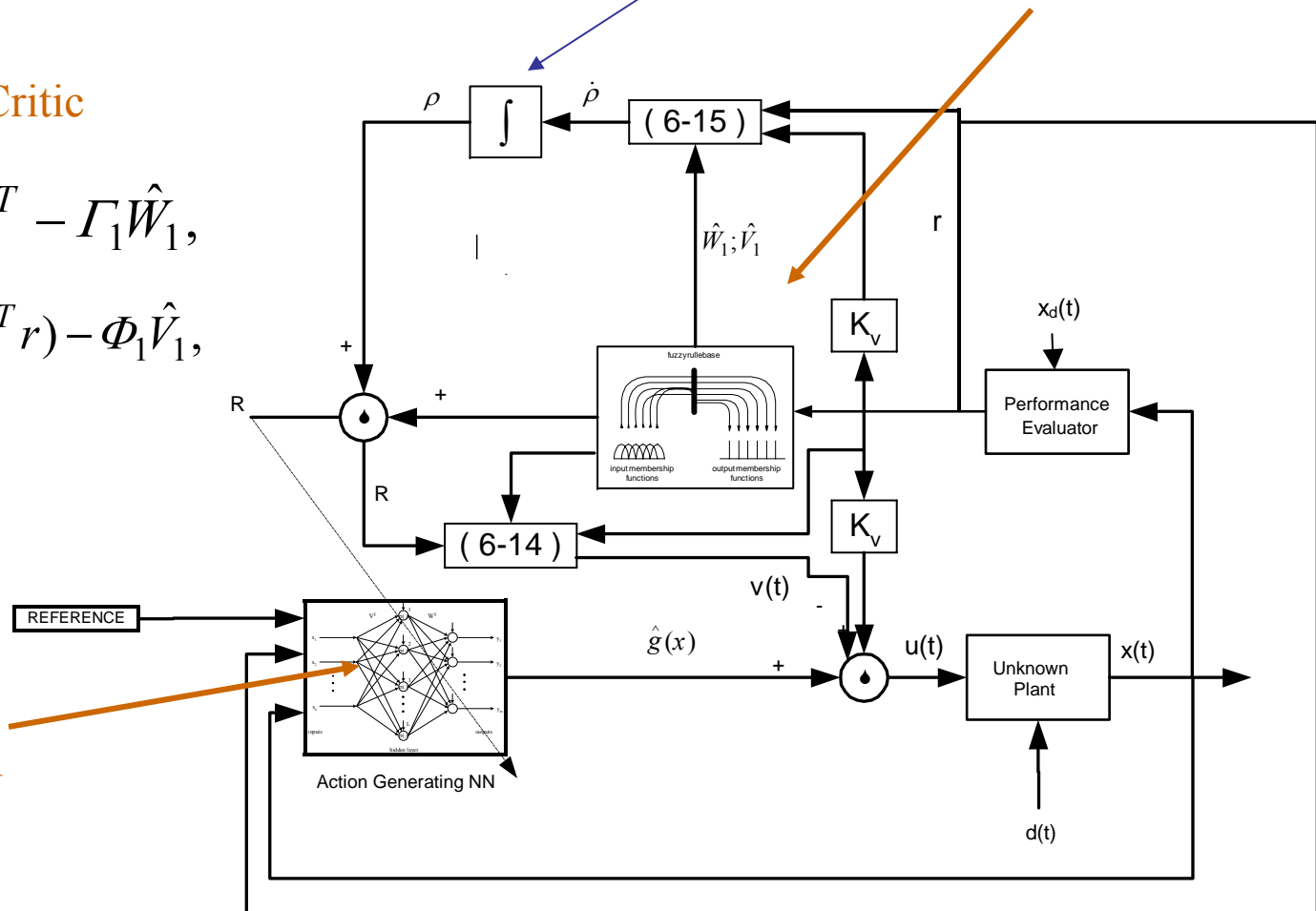
$$\dot{\hat{W}}_1 = -\mu(\hat{V}_1^T r) R^T - \Gamma_1 \hat{W}_1,$$

$$\dot{\hat{V}}_1 = -r H^T \hat{W}^T \mu'(\hat{V}_1^T r) - \Phi_1 \hat{V}_1,$$

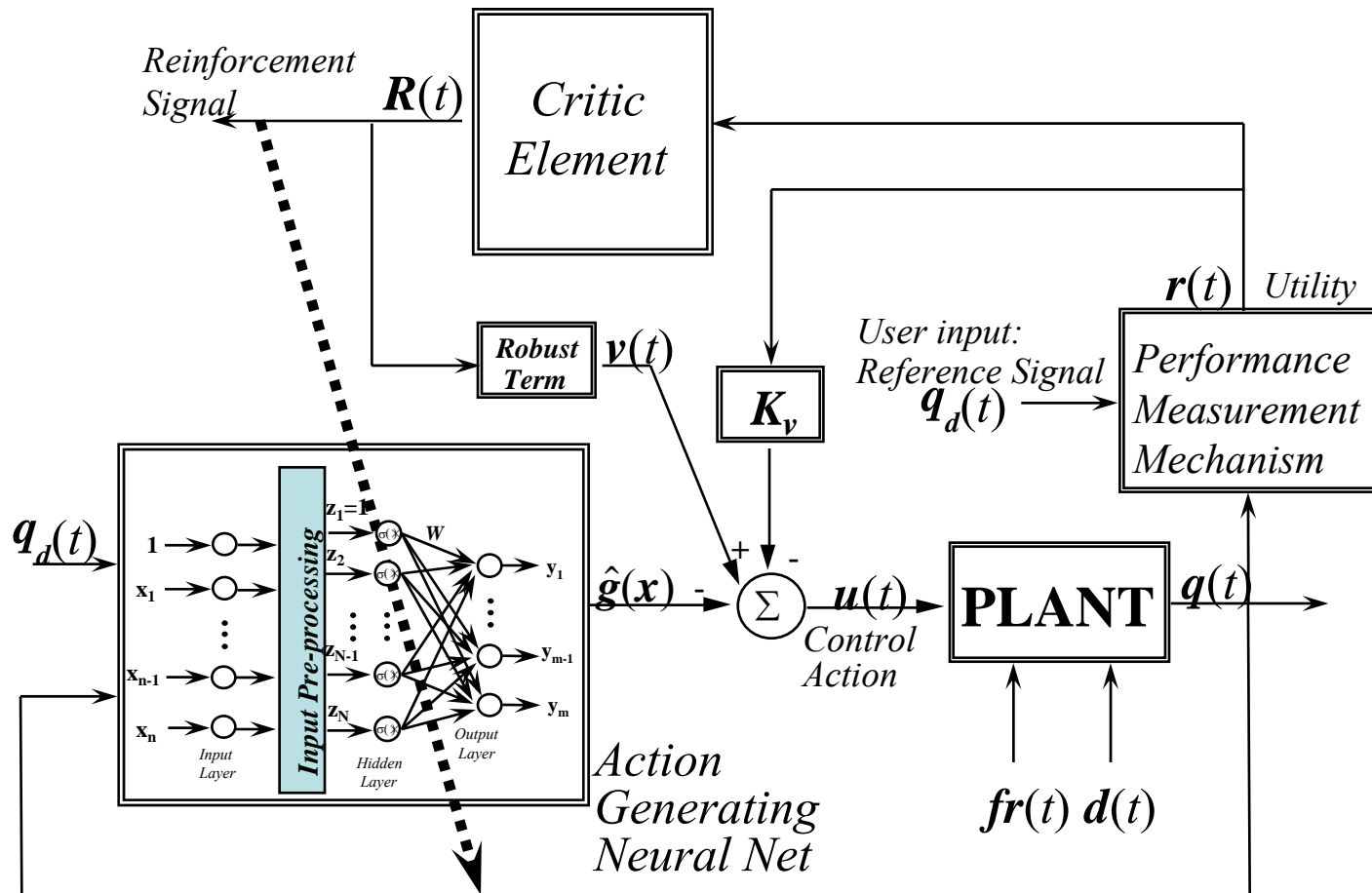
Critic requires MEMORY

FL Critic

Action generating NN



Reinforcement Learning NN Controller



High-Level NN Controllers Need Exotic Lyapunov Fns.

Reinforcement NN control

Simplified critic signal

$$R(t) = \text{sgn}(r(t)) = \pm 1$$

Lyapunov Fn

$$L(t) = \sum_{i=1}^n |r_i| + \frac{1}{2} \text{tr}(\tilde{W}^T F^{-1} \tilde{W})$$

$$\dot{L} = \text{sgn}(\mathbf{r})^T \dot{\mathbf{r}} + \text{tr}(\tilde{W}^T \mathbf{F}^{-1} \dot{\tilde{W}})$$

Lyap. Deriv. contains $R(t)$!!

Tuning Law only contains $R(t)$

$$\dot{\hat{W}} = F \sigma(x) R^T - \kappa F \hat{W}$$

Adaptive Reinforcement Learning

Critic is output of NN #1

$$R = \hat{W}_1^T \cdot \sigma(\chi_1) + \rho,$$

$$L(t) = \ln(1 + e^{-\alpha r(t)}) + \ln(1 + e^{\alpha r(t)}) + \frac{1}{2} \text{tr}(\tilde{W}^T F^{-1} \tilde{W})$$

$$\dot{L} = \left(\frac{\alpha^+}{1 + e^{-\alpha^+ r(t)}} + \frac{-\alpha^-}{1 + e^{\alpha^- r(t)}} \right) \dot{r}(t) + \text{tr}(\tilde{W}^T \mathbf{F}^{-1} \dot{\tilde{W}})$$

Action is output of second NN

$$\hat{g}(x, x_d) = \hat{W}_2^T \sigma(\chi_2)$$

The tuning algorithm treats this as a SINGLE 2-layer NN

$$\dot{\hat{W}}_1 = -\sigma(\chi_1) R^T - \hat{W}_1,$$

$$\dot{\hat{W}}_2 = \Gamma \sigma(\chi_2) \cdot \left(r + V_1 \sigma'(\chi_1)^T \hat{W}_1 R \right)^T - \Gamma \hat{W}_2,$$

Encode Information into the Value Function

Principe- Entropy

Information-Theoretic Learning

$$H(x_0, u(x, t), p(u)) = - \int \int p(x_0, u) \ln p(x_0, u) du dx_0$$

Renyi's entropy

Corentropy

Brockett- Minimum-Attention Control

awareness & effort (partial derivatives in PM)

$$V(x_0, u) = \int r(x, u) dt + \iint a \left(\frac{\partial u}{\partial t} \right)^2 + b \left(\frac{\partial u}{\partial x} \right)^2 dx dt$$

2. Neural Network Solution of Optimal Design Equations

Nearly Optimal Control

Based on HJ Optimal Design Equations

Known system dynamics

Preliminary Off-line tuning

Before-

1. Neural Networks for Feedback Control

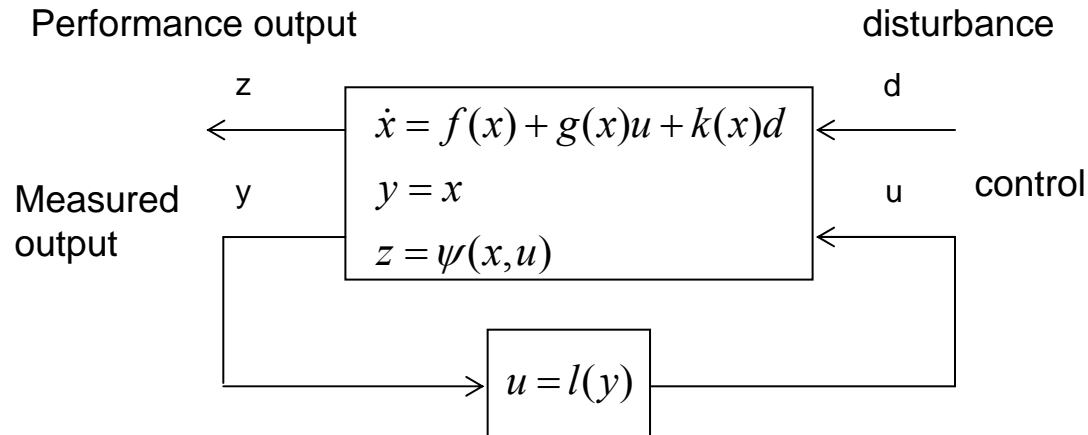
Based on FB Control Approach

Unknown system dynamics

On-line tuning

H-Infinity Control Using Neural Networks

System



L_2 Gain Problem

$$\|z\|^2 = h^T h + \|u\|^2$$

Find control $u(t)$ so that

$$\frac{\int_0^\infty \|z(t)\|^2 dt}{\int_0^\infty \|d(t)\|^2 dt} = \frac{\int_0^\infty (h^T h + \|u\|^2) dt}{\int_0^\infty \|d(t)\|^2 dt} \leq \gamma^2$$

For all L_2 disturbances
And a prescribed gain γ^2

Zero-Sum differential game

Standard Bounded L_2 Gain Problem

$$J(u, d) = \int_0^{\infty} \left(h^T h + \|u\|^2 - \gamma^2 \|d\|^2 \right) dt$$

Game theory value function

Take $\|u\|^2 = u^T R u$ and $\|d\|^2 = d^T d$

Hamilton-Jacobi Isaacs (HJI) equation

$$0 = V_x^T f + h^T h - \frac{1}{4} V_x^T g R^{-1} g^T V_x + \frac{1}{4\gamma^2} V_x^T k k^T V_x$$

Stationary Point

$$u^* = -\frac{1}{2} R^{-1} g^T(x) V_x$$

Optimal control

$$d^* = \frac{1}{2\gamma^2} k^T(x) V_x$$

Worst-case disturbance

If HJI has a positive definite solution V and the associated closed-loop system is AS then L_2 gain is bounded by γ^2

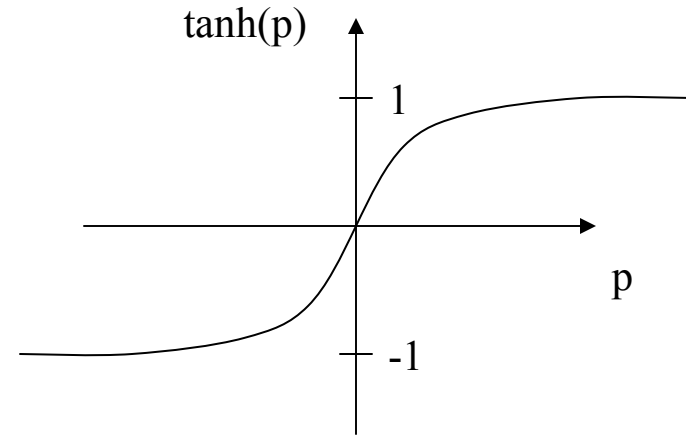
Problems to solve HJI

Beard proposed a successive solution method using Galerkin approx.

Viscosity Solution

Bounded L₂ Gain Problem for Constrained Input Systems

Control constrained by saturation function $\phi(\cdot)$



Encode constraint into Value function

$$J(u, d) = \int_0^\infty \left(h^T h + 2 \int_0^u \phi^{-T}(v) dv - \gamma^2 \|d\|^2 \right) dt$$

$$\|u\|_q^2 = 2 \int_0^u \phi^{-T}(v) dv$$

(Used by Lyshevsky for H₂ control)

This is a quasi-norm

Weaker than a norm –

homogeneity property is replaced by the weaker symmetry property $\|x\|_q = \|-x\|_q$

Hamiltonian

$$H(x, V_x, u, d) \equiv \frac{\partial V^T}{\partial x} (f + gu + kd) + h^T h + 2 \int_0^u \phi^{-T}(v) dv - \gamma^2 d^T d$$

Stationarity conditions

$$0 = \frac{\partial H}{\partial u} = g^T V_x + 2\phi^{-1}(u)$$

$$0 = \frac{\partial H}{\partial d} = k^T V_x - 2\gamma^2 d$$

Optimal inputs

$$u^* = -\frac{1}{2} \phi(g^T(x) V_x)$$

$$d^* = \frac{1}{2\gamma^2} k^T(x) V_x$$

Leibniz's Formula

Solve for $u(t)$

Note $u(t)$ is bounded!

Cannot solve HJI !!

Successive Solution- Algorithm 1:

Let γ be prescribed and fixed.

u_0 a stabilizing control with region of asymptotic stability Ω_0

1. Outer loop- update control

Initial disturbance $d^0 = 0$

2. Inner loop- update disturbance

Solve Value Equation

Consistency equation $\rightarrow \frac{\partial(V^i_j)^T}{\partial x} (f + gu_j + kd) + h^T h + 2 \int_0^{u_j} \phi^{-T}(v) dv - \gamma^2 (d^i)^T d^i = 0$

Inner loop update

$$d^{i+1} = \frac{1}{2\gamma^2} k^T(x) \frac{\partial V^i_j}{\partial x}$$

go to 2.

Iterate i until convergence to d^∞, V^∞_j with RAS Ω^∞_j

Outer loop update

$$u_{j+1} = -\frac{1}{2} \phi \left(g^T(x) \frac{\partial V^\infty_j}{\partial x} \right)$$

Go to 1.

Iterate j until convergence to $u_\infty, V^\infty_\infty$, with RAS Ω^∞_∞

CT Policy Iteration for H-Infinity Control--- c.f. Howard

Results for this Algorithm

The algorithm converges to $V^*(\Omega_0), \Omega_0, u^*(\Omega_0), d^*(\Omega_0)$
the optimal solution on the RAS Ω_0

Sometimes the algorithm converges to the optimal HJI solution V^*, Ω^*, u^*, d^*

For this to occur it is required that $\Omega^* \subseteq \Omega_0$

For every iteration on the disturbance d^i one has

$V^i_j \leq V^{i+1}_j$ the value function increases

$\Omega^i_j \supseteq \Omega^{i+1}_j$ the RAS decreases

For every iteration on the control u_j one has

$V^\infty_j \geq V^\infty_{j+1}$ the value function decreases

$\Omega^\infty_j \subseteq \Omega^\infty_{j+1}$ the RAS does not decrease

Problem- Cannot solve the Value Equation!

Neural Network Approximation for Computational Technique

Neural Network to approximate $V^{(i)}(x)$

$$V_L^{(i)}(x) = \sum_{j=1}^L w_j^{(i)} \sigma_j(x) = W_L^{T(i)} \bar{\sigma}_L(x),$$

Value function gradient approximation is

$$\frac{\partial V_L^{(i)}}{\partial x} = \frac{\partial \bar{\sigma}_L(L)^T}{\partial x} W_L^{(i)} = \nabla \bar{\sigma}_L^T(x) W_L^{(i)}$$

Substitute into Value Equation to get

$$0 = w_j^{i T} \nabla \sigma(x) \dot{x} + r(x, u_j, d^i) = w_j^{i T} \nabla \sigma(x) f(x, u_j, d^i) + h^T h + \|u_j\|^2 - \gamma^2 \|d^i\|^2$$

Therefore, **one may solve for NN weights** at iteration (i,j)

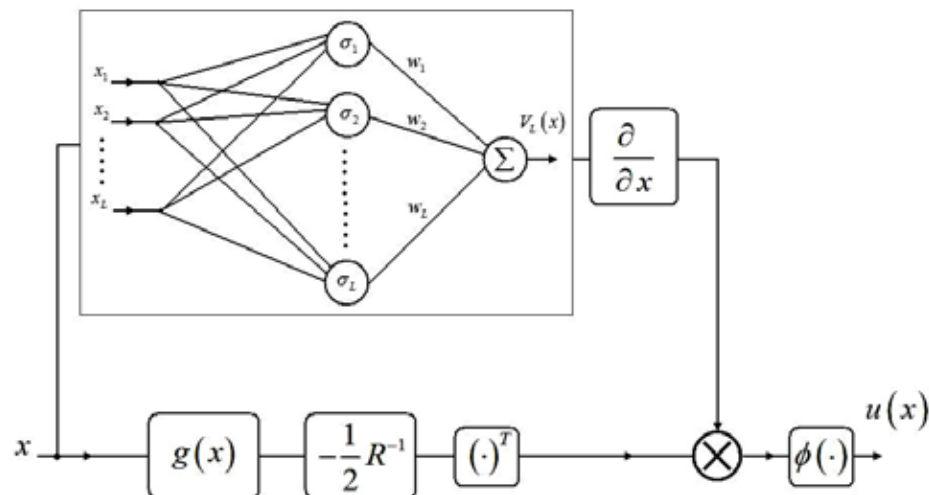
Neural Network Feedback Controller

Optimal Solution

$$d = \frac{1}{2} k^T(x) \nabla \bar{\sigma}_L^T W_L.$$

$$u = -\frac{1}{2} \phi \left(g^T(x) \nabla \bar{\sigma}_L^T W_L \right)$$

A NN feedback controller with nearly optimal weights



Example: Linear system

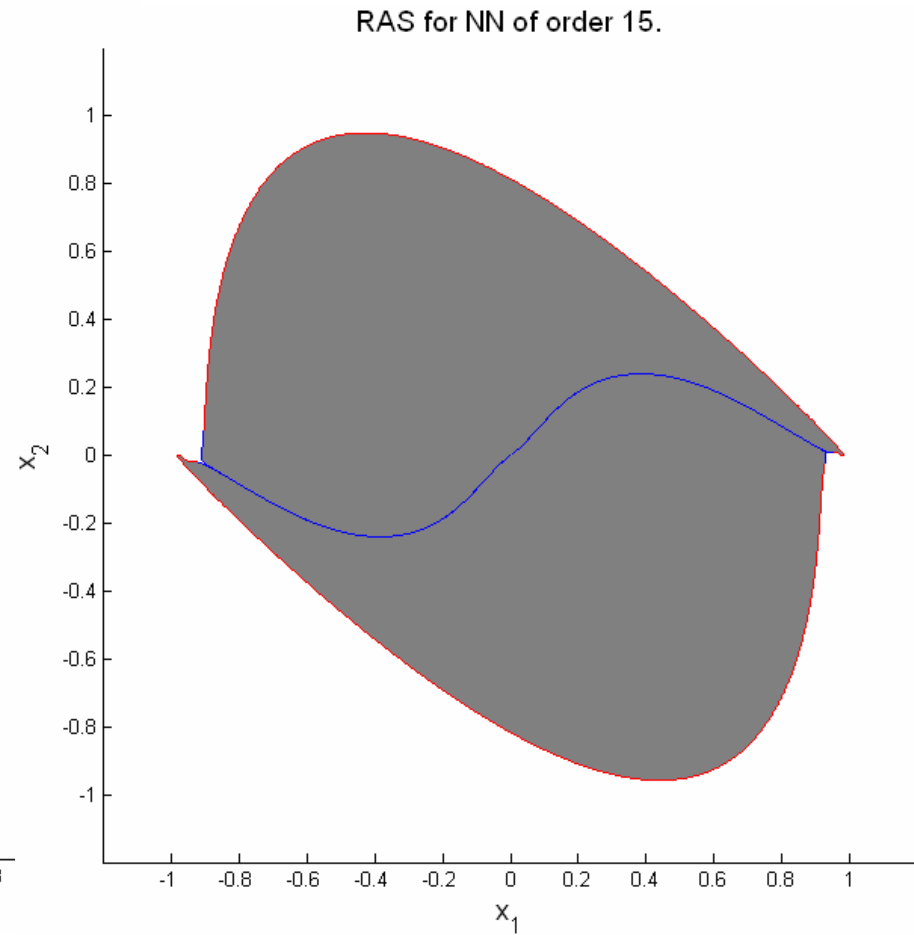
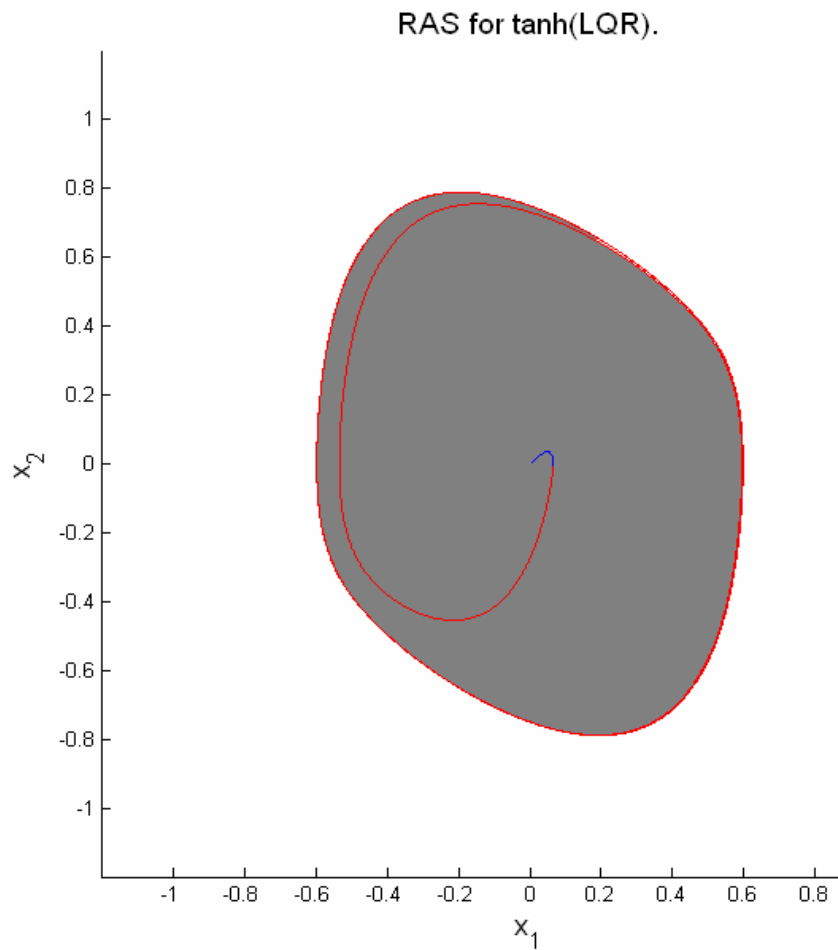
$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & -0.5 \\ 1 & 1.5 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ -1 \end{bmatrix} u, \quad |u| \leq 1$$

$$\begin{aligned} V_{15}(x_1, x_2) = & w_1 x_1^2 + w_2 x_2^2 + w_3 x_1 x_2 + w_4 x_1^4 + \\ & w_5 x_2^4 + w_6 x_1^3 x_2 + w_7 x_1^2 x_2^2 + w_8 x_1 x_2^3 + w_9 x_1^6 + w_{10} x_2^6 \\ & w_{11} x_1^5 x_2 + w_{12} x_1^4 x_2^2 + w_{13} x_1^3 x_2^3 + w_{14} x_1^2 x_2^4 + w_{15} x_1 x_2^5 \end{aligned}$$

Activation functions = even polynomial basis up to order 6

Initial Gain found by LQR

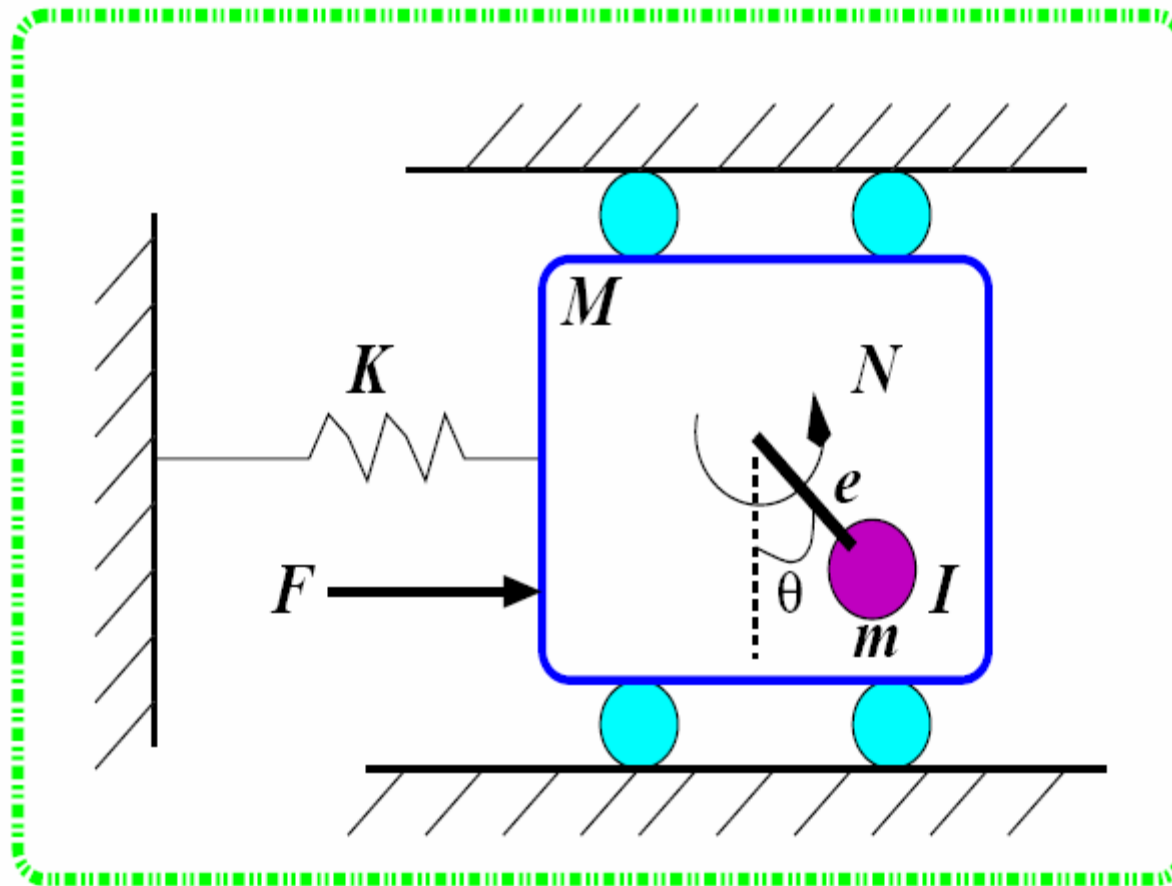
Optimal NN solution



RAS found by integrating $\dot{x} = -f(x)$

That is, reverse time $dt = -d\tau$

Rotational-Translational Actuator Benchmark Problem



F is a disturbance

Control input is torque N

Rotational-Translational Actuator Benchmark Problem

$$\dot{x} = f(x) + g(x)u + k(x)d$$

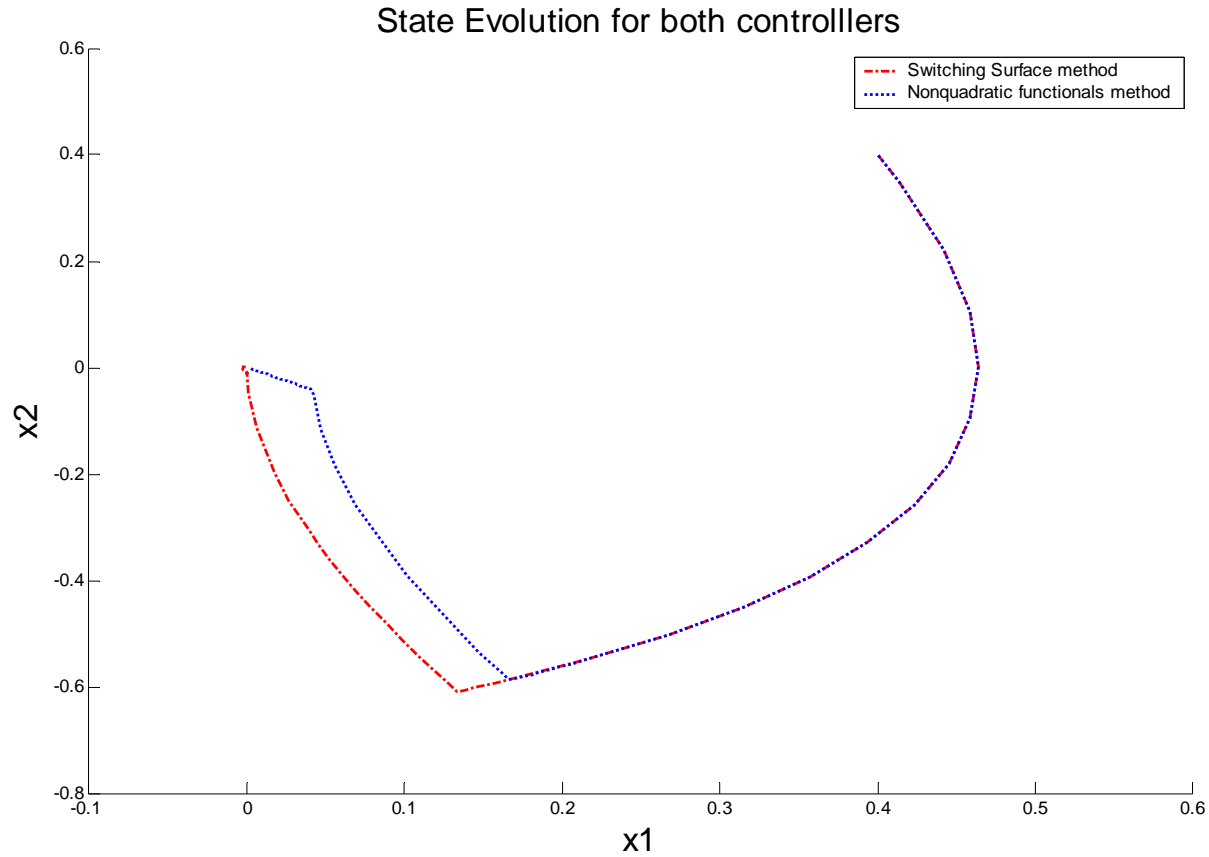
$$f(x) = \begin{bmatrix} x_2 \\ \frac{-x_1 + \varepsilon x_4^2 \sin x_3}{1 - \varepsilon^2 \cos^2 x_3} \\ x_4 \\ \frac{\varepsilon \cos x_3 (x_1 - \varepsilon x_4^2 \sin x_3)}{1 - \varepsilon^2 \cos^2 x_3} \end{bmatrix}, \quad g(x) = \begin{bmatrix} 0 \\ \frac{-\varepsilon \cos x_3}{1 - \varepsilon^2 \cos^2 x_3} \\ 0 \\ \frac{1}{1 - \varepsilon^2 \cos^2 x_3} \end{bmatrix}$$

$$\varepsilon = 0.2$$

Minimum-Time Control

Encode into Value Function

$$V = \int_0^{\infty} \left[\tanh(x^T Q x) + 2 \int_0^u (\phi^{-1}(\mu))^T R d\mu \right] dt$$



3. Approximate Dynamic Programming

Nearly Optimal Control

Based on recursive equation for the optimal value

Usually Known system dynamics (except Q learning)

The Goal – unknown dynamics

On-line tuning

Before-

2. Neural Network Solution of Optimal Design Equations

Nearly Optimal Control

Based on HJ Optimal Design Equations

Known system dynamics

Preliminary Off-line tuning

1. Neural Networks for Feedback Control

Based on FB Control Approach

Unknown system dynamics

On-line tuning

IEEE Trans. Neural Networks
Special Issue on Neural Networks for Feedback Control

Lewis, Wunsch, Prokhorov, Jie Huang, Parisini

Due date 1 December

Bring together:

Feedback control system community

Approximate Dynamic Programming community

Neural Network community

Discrete-Time Systems

$$x_{k+1} = f(x_k, u_k) \quad V(x_k) = \sum_{i=k}^N \gamma^{i-k} r(x_k, u_k)$$

Value in difference form -

$$V_h(x_k) = r(x_k, h(x_k)) + \gamma V_h(x_{k+1})$$

**Recursive form
Consistency equation**

Howard Policy Iteration- Iterate the following until convergence

1. Find the value for the prescribed policy

$$V_j(x_k) = r(x_k, h_j(x_k)) + \gamma V_j(x_{k+1})$$

solve completely

2. Policy improvement

$$h_{j+1}(x_k) = \arg \min_{u_k} (r(x_k, u_k) + \gamma V_j(x_{k+1}))$$

Four ADP Methods proposed by Werbos

Critic NN to approximate:

Heuristic dynamic programming

$$\text{Value } V(x_k)$$

AD Heuristic dynamic programming
(Watkins Q Learning)

$$\text{Q function } Q(x_k, u_k)$$

Dual heuristic programming

$$\text{Gradient } \frac{\partial V}{\partial x}$$

AD Dual heuristic programming

$$\text{Gradients } \frac{\partial Q}{\partial x}, \frac{\partial Q}{\partial u}$$

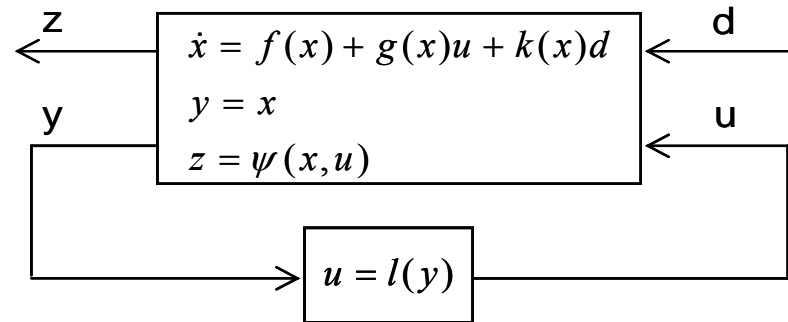
Action NN to approximate the Control

Bertsekas- Neurodynamic Programming

Barto & Bradtke- Q-learning proof (Imposed a settling time)

Continuous-Time Systems

$$V(x(t)) = \int_t^T r(x, u, d) dt$$



Value in differential form -

$$0 = \left(\frac{\partial V}{\partial x} \right)^T (f + gu + kd) + r(x, u, d) \equiv H(x, \frac{\partial V}{\partial x}, u, d) \quad \text{Consistency equation}$$

$$V_h(x_k) = r(x_k, h(x_k)) + \gamma V_h(x_{k+1})$$

$$u^*(x(t)) = -\frac{1}{2} g^T(x) \frac{\partial V^*}{\partial x} \qquad d^*(x(t)) = \frac{1}{2\gamma^2} k^T(x) \frac{\partial V^*}{\partial x}$$

HJB equation

$$0 = \left(\frac{dV^*}{dx} \right)^T f + h^T h - \frac{1}{4} \left(\frac{dV^*}{dx} \right)^T g g^T \frac{dV^*}{dx} + \frac{1}{4\gamma^2} \left(\frac{dV^*}{dx} \right)^T k k^T \frac{dV^*}{dx}$$

Continuous Time Policy Iteration

Select a stabilizing initial control

Abu-Khalaf and Lewis- H inf

1. Outer loop- update control

Initial disturbance set to zero

Saridis – H_2

2. Inner loop- update disturbance

Solve Lyapunov equation

$$\frac{\partial(V_j^i)^T}{\partial x} (f + gu_j + kd^i) + h^T h + \|u_j\|^2 - \gamma^2 \|d^i\|^2 = 0$$

Inner loop disturbance update

$$d^{i+1} = \frac{1}{2\gamma^2} k^T(x) \frac{\partial V_j^i}{\partial x}$$

go to 2.

Until convergence

c.f. Howard work in DT Systems

Outer loop update

$$u_{j+1} = -\frac{1}{2} \left(g^T(x) \frac{\partial V_j^i}{\partial x} \right)$$

Go to 1.

Until convergence

Neural Network Approximation of Value Function

$$\hat{V}(x, w^i_j) = w_j^{iT} \sigma(x)$$

Lyapunov equation becomes

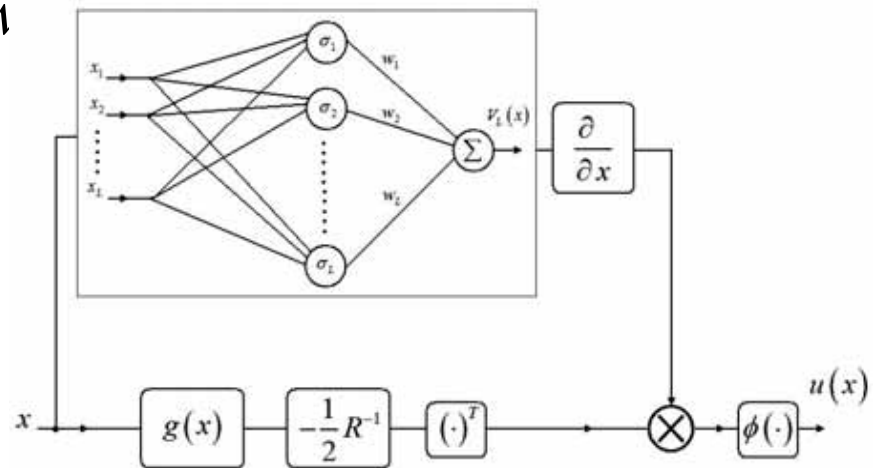
$$0 = w_j^{iT} \nabla \sigma(x) \dot{x} + r(x, u_j, d^i) = w_j^{iT} \nabla \sigma(x) f(x, u_j, d^i) + h^T h + \|u_j\|^2 - \gamma^2 \|d^i\|^2$$

Control action

$$u^*(x) = -\frac{1}{2} R^{-1} g^T(x) \nabla \sigma^T(x) w^*$$

CT Approx Policy Iteration
Abu-Khalaf & Lewis

Nearly optimal FB control
Off-line tuning
Known dynamics



CT Nearly Optimal NN feedback

Continuous-time adaptive critic

Abu-Khalaf & Lewis
(c.f. Doya)

On-line tuning

Critic NN $V(x) = w^T \sigma(x)$

Hamiltonian (CT consistency check)

$$H(x, \frac{\partial V}{\partial x}, u) = \dot{V} + r(x, u) = \left(\frac{\partial V}{\partial x} \right)^T \dot{x} + r(x, u) = \left(\frac{\partial V}{\partial x} \right)^T f(x, u) + r(x, u) = 0$$

residual eq error

$$\delta = \frac{dw^T \sigma}{dt} + r(x, u) = w^T \nabla \sigma(x) \dot{x} + r(x, u) = w^T \nabla \sigma(x) f(x, u) + r(x, u)$$

$$E = \frac{1}{2} |\delta|^2$$

Target value

$$\frac{\partial E}{\partial w} = \delta(t) \frac{\partial \delta}{\partial w} = \delta(t) \nabla \sigma(x) f(x, u) \quad \text{gradient}$$

$$\dot{w} = -\alpha \nabla \sigma(x) f(x, u) \delta$$

Update weights using, e.g., gradient descent
Or RLS

Action NN

$$Y_2 = -\frac{1}{2} R^{-1} g^T(x) \nabla \sigma^T(x) w = \bar{\phi}^T w$$

Critic weights

Target action

$$\bar{\phi}^T(x) = -\frac{1}{2} R^{-1} g^T(x) \nabla \sigma^T(x) \quad \text{Activation fns depend on system dynamics}$$

$$\hat{Y}_2 = \bar{\phi}^T(x) v \quad \text{Action NN}$$

$$e_2(x) = \hat{Y}_2 - Y_2 = -\frac{1}{2} R^{-1} g^T(x) \nabla \sigma^T(x) [v - w] = \bar{\phi}^T(x) [v - w]$$

$$\dot{v} = -\beta \bar{\phi}^T(x) e_2(x) \quad \text{update weights by gradient descent}$$

Alternative, simply set $u(x) = Y_2 = -\frac{1}{2} R^{-1} g^T(x) \nabla \sigma^T(x) w = \bar{\phi}^T w$

Does not work- proof development so far indicates that
critic NN must be tuned faster than action NN
i.e. $\alpha > \beta$

c.f. Bradtke & Barto DT Q learning work

Small Time-Step Approximate Tuning for Continuous-Time Adaptive Critics

Sampled data systems

$$H(x, \frac{\partial V}{\partial x}, u) = \dot{V}(x) + r(x, u) \approx \frac{V_{t+1} - V_t}{\Delta t} + r(x, u) \approx \frac{V_{t+1} - V_t}{\Delta t} + \frac{r^D(x_t, u_t)}{\Delta t}$$

$$A_1^*(x_t, u_t) = \frac{r^D(x_t, u_t) + V(x_{t+1}) - V^*(x_t)}{\Delta t}$$

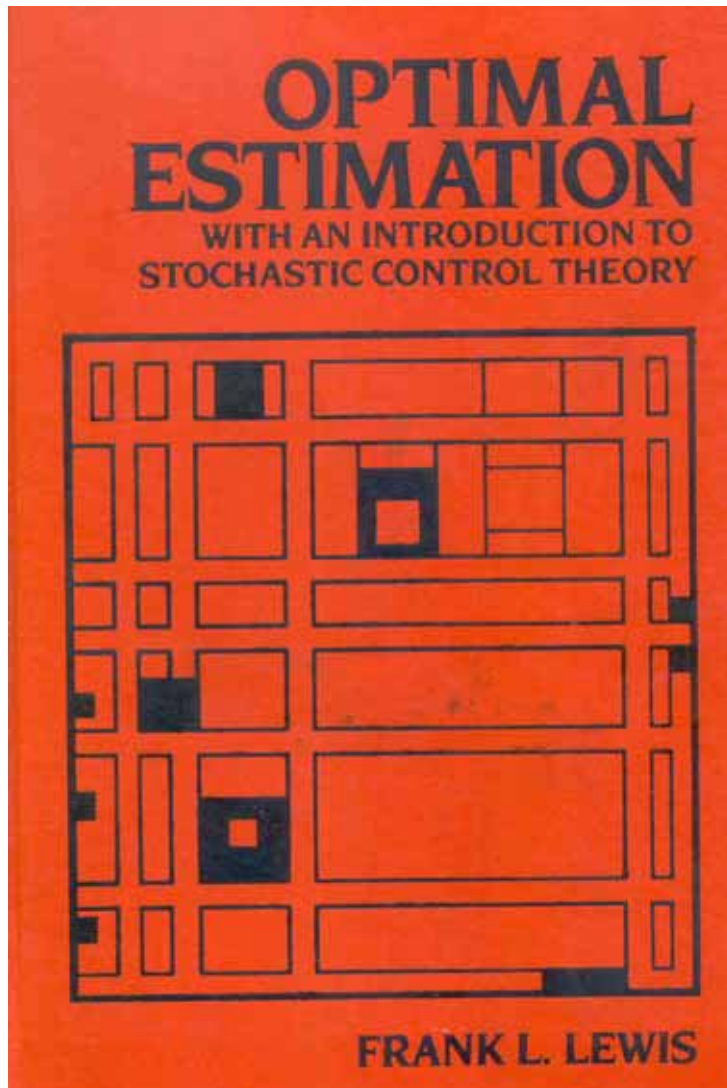
Baird's Advantage function

This is not in standard DT form

$$V_h(x_k) = r(x_k, h(x_k)) + \gamma V_h(x_{k+1})$$

For More Information

Journal papers on <http://arri.uta.edu/acs>



Optimal Control
Lewis & Syrmos 1995

BRIAN L. STEVENS and FRANK L. LEWIS



**AIRCRAFT
CONTROL
AND
SIMULATION**



*second
edition*

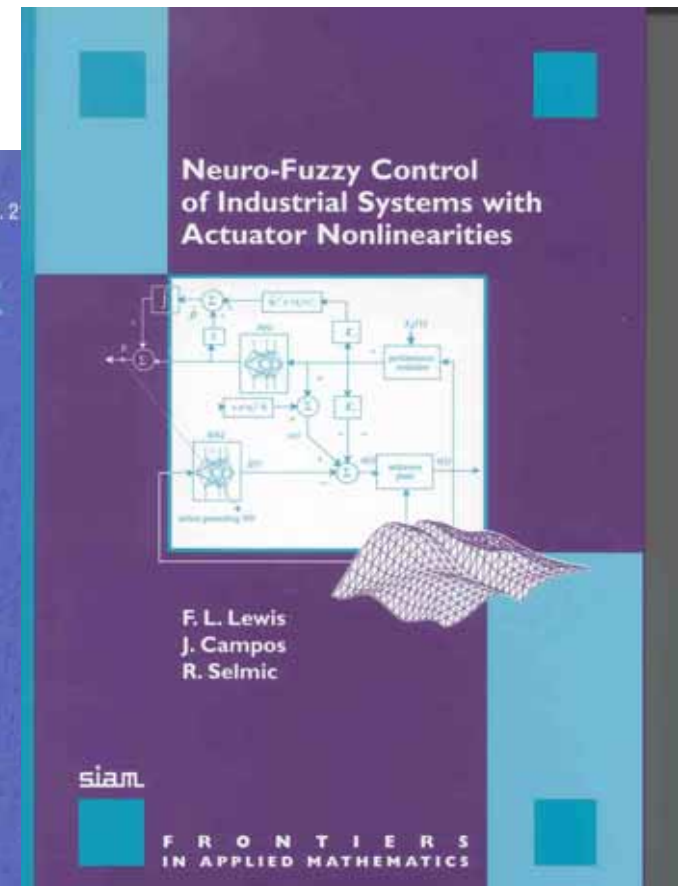
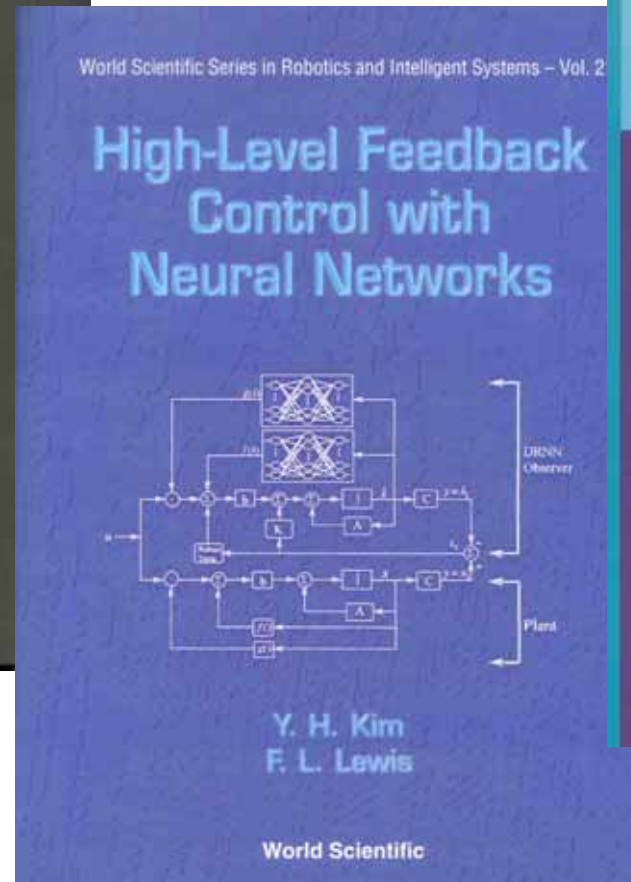
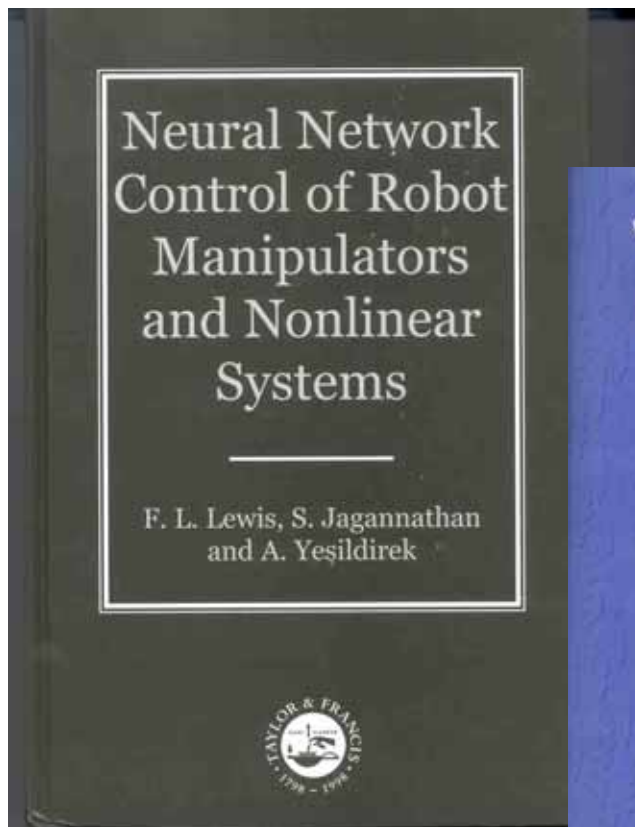
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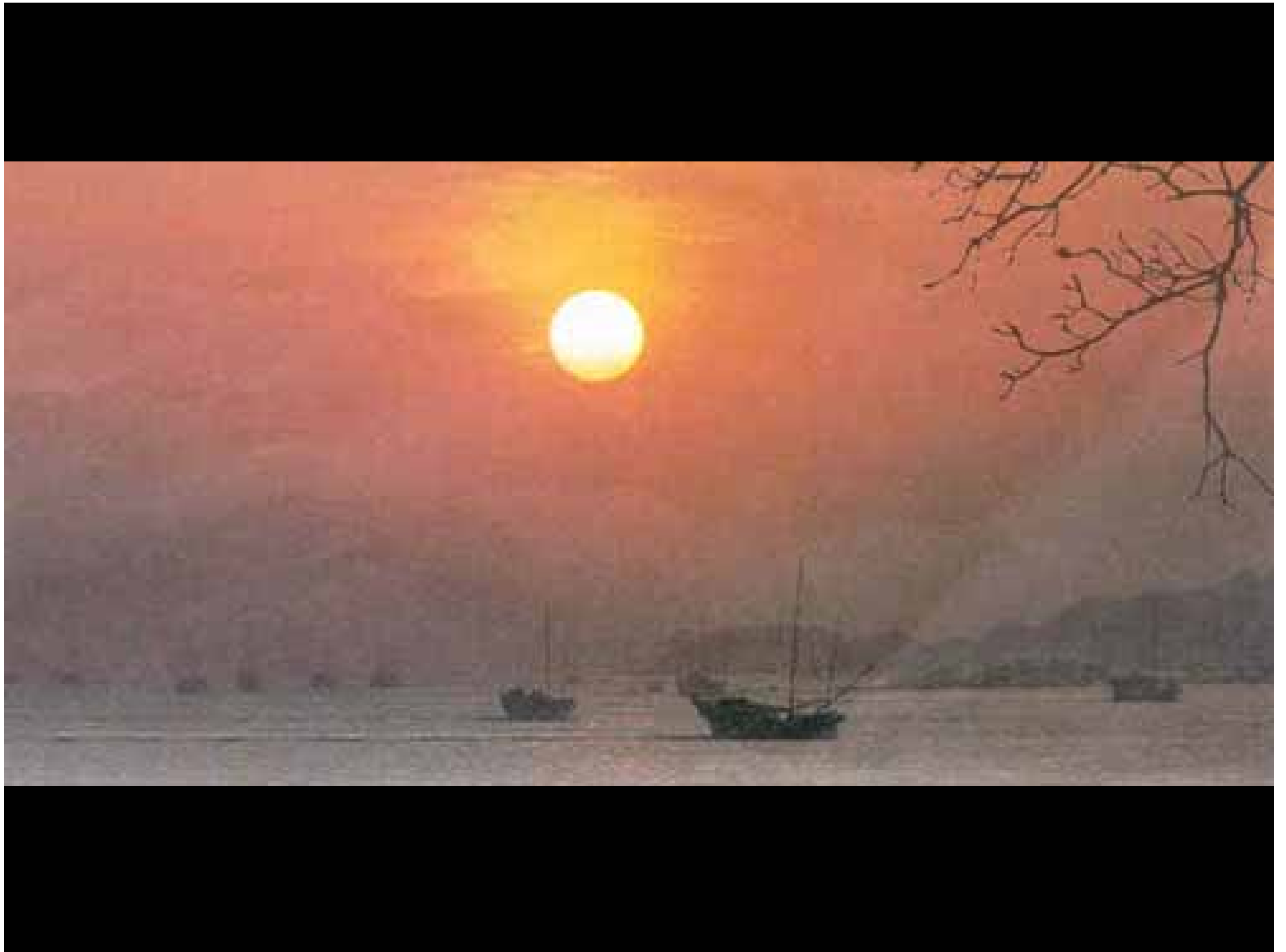
Second Edition, Revised and Expanded



Frank L. Lewis
Darren M. Dawson
Chaouki T. Abdallah



**In Progress: M. Abu-Khalaf, Jie Huang, F.L. Lewis
Nearly Optimal Control by HJ Equation Solution Using Neural Networks**



Theorem 1.

Necessary and Sufficient Conditions for H-infinity Static OPFB Control

Assume that $Q > 0$, then system (1) is output-feedback stabilizable with L_2 gain bounded by γ if and only if:

- i. (A, C) is detectable
- ii. There exist matrices K^* and L such that

$$K^* C = R^{-1} (B^T P + L)$$

where $P > 0$, $P^T = P$, is a solution of

$$PA + A^T P + Q + \frac{1}{\gamma^2} P D D^T P - P B R^{-1} B^T P + L^T R^{-1} L = 0$$

ONLY TWO COUPLED EQUATIONS c.f. results by Kucera and De Souza

Note there is an (A, B) stabilizability condition hidden in the existence of Solution to the Riccati eq.

Solution Algorithm 1- c.f. Geromel

1. Initialize:

Set $n=0$, $L_0 = 0$, and select γ , Q, R

2. n -th iteration:

solve for P_n in the ARE

$$P_n A + A^T P_n + Q + \frac{1}{\gamma^2} P_n D D^T P_n - P_n B R^{-1} B^T P_n + L_n^T R^{-1} L_n = 0$$

Evaluate gain and update L

$$K_{n+1} = R^{-1} (B^T P_n + L) C^T (C C^T)^{-1}$$

$$L_{n+1} = R K_{n+1} C - B^T P_n$$

Until Convergence

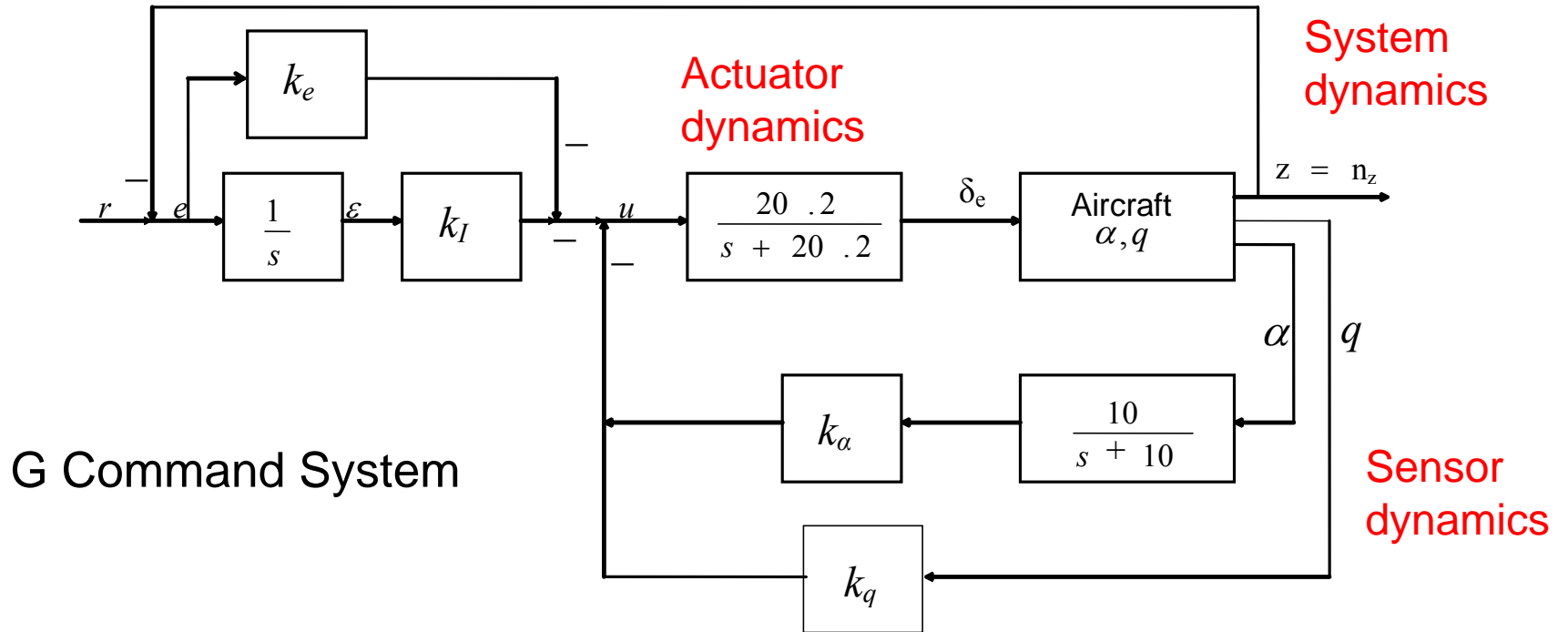
Based on ARE, so no initial stabilizing gain needed !!

Tries to project gain onto nullspace perp. of C using degrees of freedom in L

Aircraft Autopilot Design



F-16 Normal Acceleration Regulator Design



$$y = [\alpha_F \quad q \quad e \quad \varepsilon]^T$$

$$u = -Ky = -[k_\alpha \quad k_q \quad k_e \quad k_I]y$$

Theorem 2. - new work

Parametrization of all H-infinity Static SVFB Controls

Assume that $Q > 0$, then K is a stabilizing SVFB with L_2 gain bounded by γ if and only if:

i. (A, B) is stabilizable

ii. There exist a matrix L such that

$$K = R^{-1}(B^T P + L)$$

where $P > 0$, $P^T = P$, is a solution of

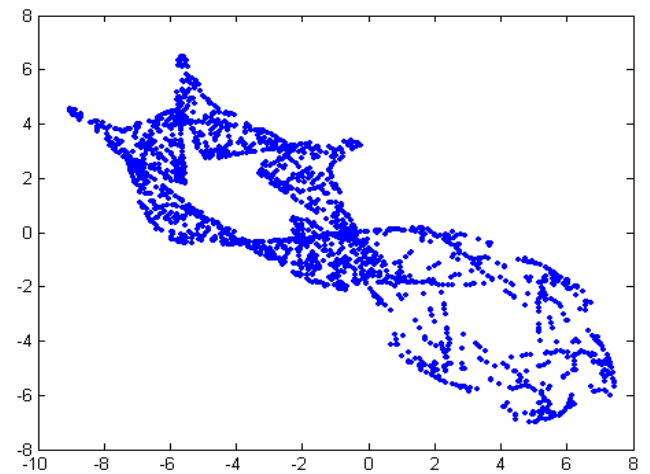
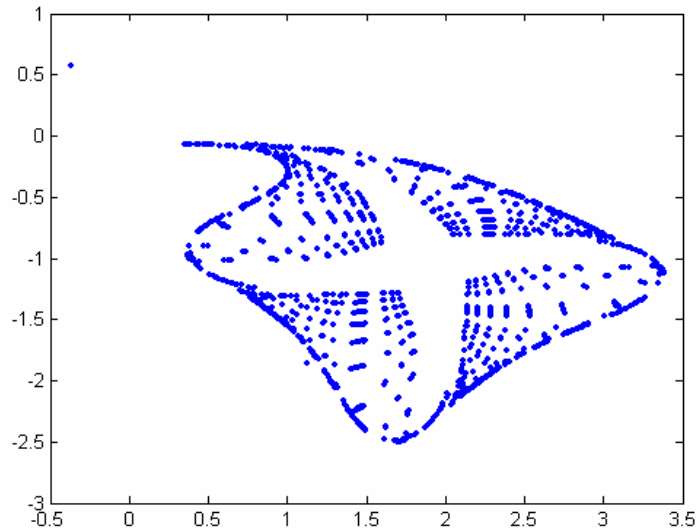
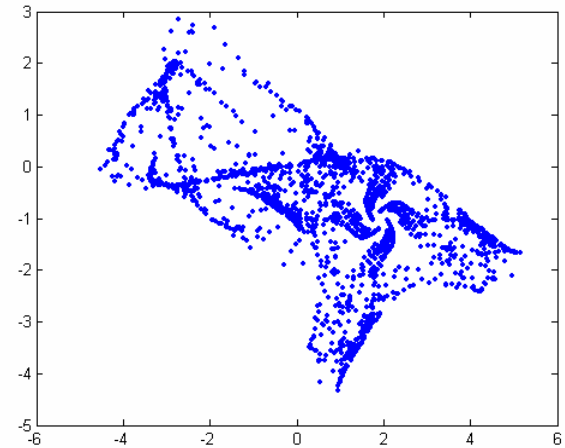
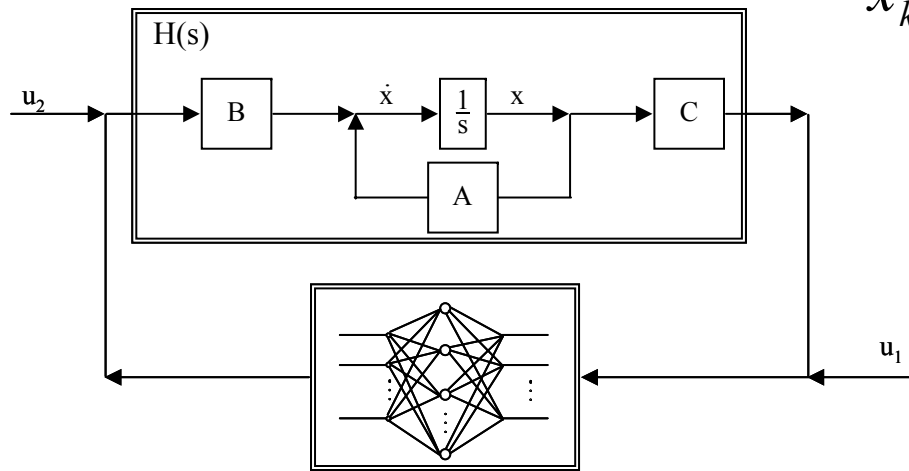
$$PA + A^T P + Q + \frac{1}{\gamma^2} P D D^T P - P B R^{-1} B^T P + L^T R^{-1} L = 0$$

OPFB is a special case

Chaos in Dynamic Neural Networks

c.f. Ron Chen

$$x_{k+1} = Ax_k + W^T \sigma(V^T x_k) + u_k$$



Jun Wang

$$z_{k+1} = \beta z_k$$

$$y_{k+1} = \alpha y_k + g - z_k \left(\frac{1}{1 + e^{-y_k / \rho}} - I \right)$$

%MATLAB file for chaotic NN
from **Jun Wang's** paper

```
function [ki,x,y,z]=tcnn(N);  
y(1)= rand; ki(1)=1; z(1)= 0.08;  
a=0.9; e= 1/250; Io=0.65;  
g= 0.0001; b=0.001;
```

```
for k=1: N-1;  
    ki(k+1)= k+1;  
    x(k)= 1/(1+exp(-y(k)/e));  
    y(k+1)= a*y(k) + g -  
    z(k)*(x(k) - Io);  
    z(k+1)= (1-b)*z(k);  
end  
x(N)= 1/(1+exp(-y(N)/e));
```

